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FORTRAN SIN

**A One-Dimensional Hydrodynamic Code
for Problems Which Include Chemical Reactions,
Elastic-Plastic Flow, Spalling, and Phase Transitions**



UNITED STATES
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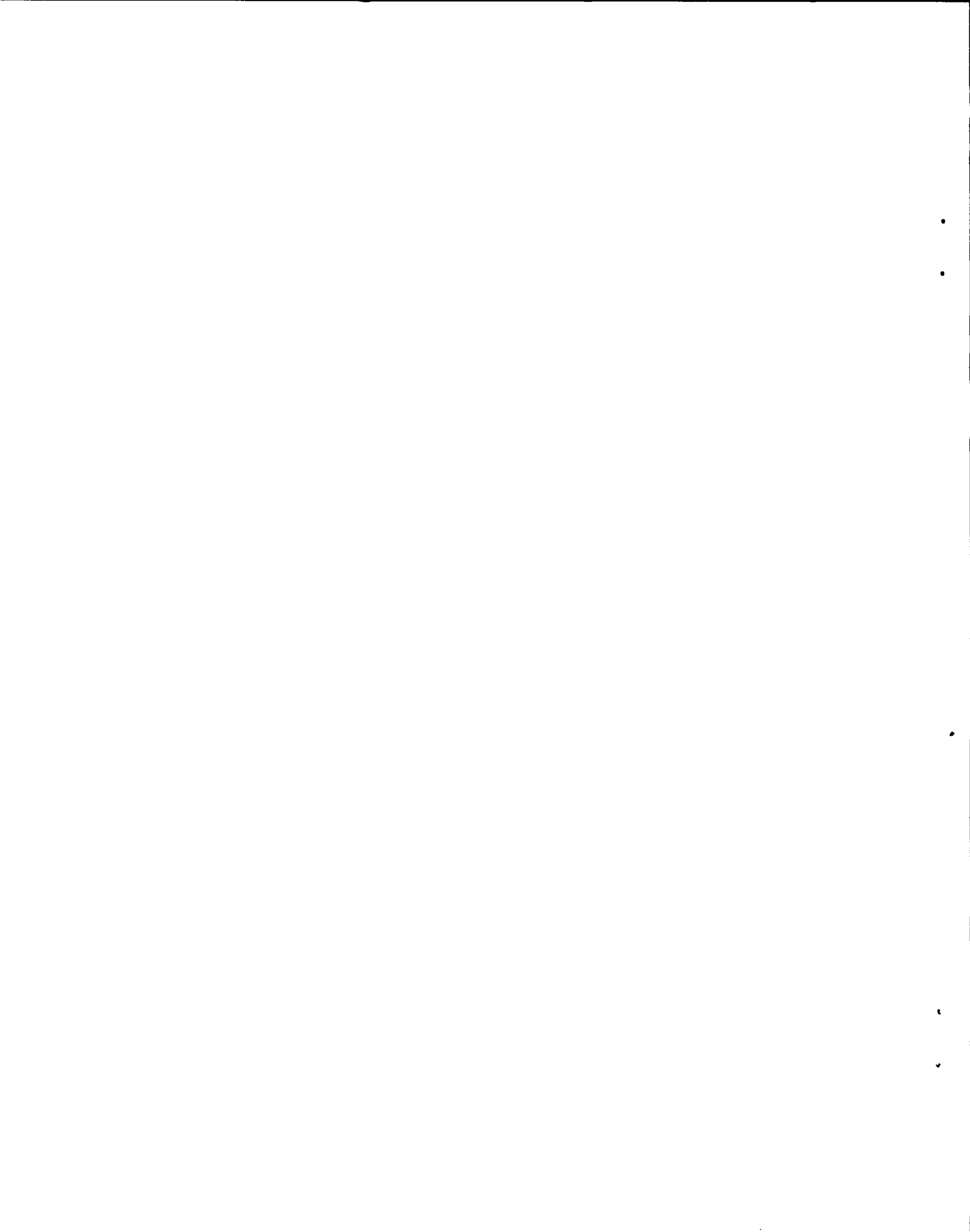


by

Charles L. Mader

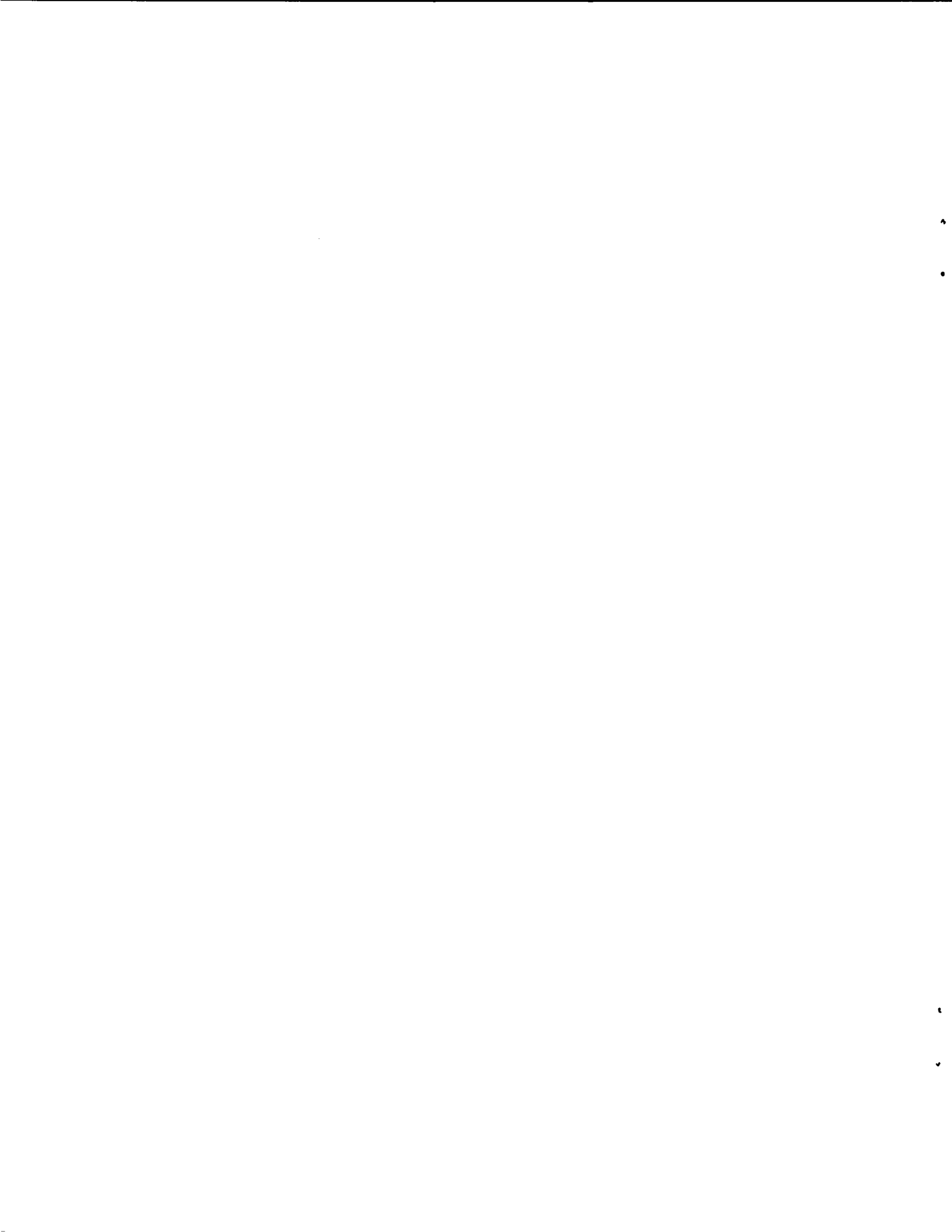
William R. Gage





CONTENTS

	Page
Abstract	5
I. Introduction	5
II. The Flow Equations	6
The Nomenclature	6
The Differential Equations	6
The Difference Equations	7
A. The Initial Conditions	7
B. The Conservation Equations for Time Advancement	7
C. Viscosity	7
D. Burn Techniques	8
E. The Elastic Stress Deviators	8
F. The Equation of State and Yield Calculation	8
G. Total Energy of Components	9
H. The Boundary Conditions	9
III. The HOM Equation of State Subroutine	10
The Nomenclature	10
The Method	10
A. Condensed Components	10
B. Gas Components	11
C. Mixture of Condensed and Gaseous Components	11
The Calling Sequence	11
Appendix A. FORTRAN SIN Input	12
Appendix B. The FORTRAN SIN Code	14
Variables Used in FORTRAN SIN	14
Listing of the FORTRAN Code	16
Appendix C. The Gamma-Law Equation of State and Hugoniot Temperature Programs	47
The Gamma-Law Equation of State Program	47
A. The Nomenclature	47
B. The Equations	47
C. The Input to the Code	47
D. The FORTRAN Code	47
The Hugoniot Temperature Program	53
A. The Nomenclature	53
B. The Equations	53
C. The Input to the Code	53
D. The FORTRAN Code	54
Acknowledgments	64
References	64



FORTRAN SIN: A ONE-DIMENSIONAL HYDRODYNAMIC CODE FOR PROBLEMS WHICH INCLUDE
CHEMICAL REACTIONS, ELASTIC-PLASTIC FLOW, SPALLING, AND PHASE TRANSITIONS

by

Charles L. Mader and William R. Gage

ABSTRACT

This report describes a CDC 6600 computer FORTRAN code for computing one-dimensional hydrodynamic problems in slab, cylindrical, or spherical geometry using realistic equations of state. Features available in the code include chemical reaction using an Arrhenius rate law, the C-J volume burn, or, for slabs, a gamma-law Taylor wave; elastic-plastic flow using the Hooke's law-Von Mises yield model; and spalling using the Whiteman and Skidmore model of the tensile stress at spalling as a linear function of the square root of the stress rate. The HOM equation of state is used to compute the equation of state for detonation products, undecomposed explosives, mixtures of the two, and condensed components which may have an instantaneous phase change.

I. INTRODUCTION

The SIN technique for solving reactive one-dimensional problems has been a useful research tool for almost ten years. The first version was written in machine language for the IBM 7090.¹ The second version was written in machine language for the IBM 7030 (STRETCH) and is called STRETCH SIN.² Because of its detailed treatment of the equation of state and its ability to compute the detailed structure of complicated reactive and nonreactive hydrodynamic problems, SIN has been used to study other interesting engineering and research problems. The new numerical techniques which resulted from these studies have been included in this new version of SIN, written in FORTRAN IV language for the CDC 6600 and called FORTRAN SIN.

The code includes the effects of heat conduction, viscosity, elastic-plastic flow, spalling, and reactive flow. As presented in this report, the code can handle 10 components and 999 mesh points; however, these limits can be increased several times

without exceeding the core memory of the CDC 6600 or IBM 7030. The boundary conditions may be continuous, a free surface, a piston with a constant, a linearly varying velocity, or, for slabs, a reactive steady-state velocity.³ The PIC or Landshoff forms for the artificial viscosity may be used, and a form for real viscosity in slab geometry is available.

The HOM equation of state is used. One may compute the equation of state of condensed explosives, detonation products, and mixtures of the two. The equation of state for the condensed component may include an instantaneous phase change such as that described for iron.⁴

The code includes the Hooke's law-Von Mises model for elastic-plastic flow as described for aluminum.⁵ It also includes spalling, using the Whiteman and Skidmore model of the tensile stress at spalling as a linear function of the square root of the tensile stress.^{6,7} It permits chemical reaction using an Arrhenius rate law, the C-J volume burn, or, for

slabs, a gamma-law Taylor wave.

The code includes features permitting microfilm listings and graphs of pressure, temperature, mass fraction, density, and particle velocity vs. Eulerian radius. Tape dumps are taken at input intervals of the cycle number, and the problem may be restarted from any desired tape dump. Problems with pistons may use the variable number of cells feature for a considerable reduction of running time.

The SIN code has been written not as a special purpose production code but as a general purpose research and engineering tool. It will continue to change (hopefully for the better) as new developments in the numerical description of real materials under high impulse become available.

This report describes the numerical methods of SIN in complete detail. It also presents sufficient details to enable a coder to follow and change the code. The latter information is not of interest to the casual reader.

II. THE FLOW EQUATIONS

The Nomenclature

D_{CJ}	the C-J detonation velocity
E	total energy (Mbar-cc/g)
E^*	activation energy
I	internal energy (Mbar-cc/g)
j	net point of the Lagrangian mesh
K	a constant of about 2 or the coefficient of viscosity
M	mass
n	time cycle
P	pressure (Mbar)
q	artificial viscosity
R	Eulerian radius
r	Lagrangian radius
S_x	elastic stress deviator in x or r direction
S_z	elastic stress deviator in z direction
T	temperature ($^{\circ}K$)
U	particle velocity (cm/ μ sec)
V	specific volume (cc/g)
V_{CJ}	the C-J volume of the detonation products

W	mass fraction of undecomposed explosive
Y_0	yield strength
Z	frequency factor
α	=1 for slabs, =2 for cylinders, =3 for spheres
γ	γ -law gas constant since $\ln P = A + \gamma \ln V$, $\gamma = B$ in HOM equation of state subroutine.
Δt	time (μ sec)
λ	thermal conductivity coefficient
μ	shear modulus
ρ	= $1/V$
ρ_0	$1/V_0$ where V_0 is initial specific volume
$(\rho_0)_i$	initial density for the i th component (g/cc)

The Differential Equations

The Lagrangian conservation equations in one dimension for slabs, cylinders, and spheres are:

$$\frac{\partial U}{\partial t} = -R^{\alpha-1} \frac{\partial \sigma}{\partial M} - (\alpha-1) \frac{V\varphi}{R} \quad \text{Conservation of momentum,}$$

$$V = R^{\alpha-1} \frac{\partial R}{\partial M} \quad \text{Conservation of mass,}$$

and

$$\frac{\partial E}{\partial t} = \frac{-\partial \sigma U R^{\alpha-1}}{\partial M} + \lambda \frac{\partial}{\partial M} \left(R^{\alpha-1} \frac{\partial T}{\partial R} \right) \quad \text{Conservation of energy with heat conduction term,}$$

where $E = I + 0.5U^2$.

$$\frac{\partial R}{\partial t} = U,$$

and

$$dM = \rho_0 r^{\alpha-1} dr = \rho R^{\alpha-1} dR,$$

where dM is element of mass per unit of solid angle ($\alpha=2$ or 3) or of surface ($\alpha=1$). σ is the sum of the viscous pressure, the equation of state pressure, and the stress deviator, S_x . $\varphi = 2S_x + S_z$, which for spherical geometry is $3/2 S_x$ since $S_z = -1/2 S_x$.

$$\frac{\partial S_x}{\partial t} = 2\mu \left(-\frac{\partial U}{\partial R} + \frac{1}{3V} \frac{\partial V}{\partial t} \right).$$

$$\frac{\partial S_z}{\partial t} = + \frac{2}{3} \left(\frac{\mu}{V} \frac{\partial V}{\partial t} \right).$$

In this report we use the convention that the stress deviators have the same sign as pressure; that is, positive in compression and negative in tension. This convention is the reverse of that used by most workers in the field.

The Difference Equations

The difference equations presented below are discussed in References 1 and 5. The pressure, temperature, energy, specific volume, and mass fraction of the cells are considered to be located at the centers of mass of the elements, and the particle velocity is considered to be located at the boundary between the cells.

A. The Initial Conditions

$$R_{j+\frac{1}{2}} = \sum_{i=1}^j (\Delta R)_i,$$

where $j = 1, 2, \dots$, and the i th component is located between j and $j+k$ where k is the number of cells for each component.

$$M_j = (\rho_0)_i \left[(R_{j-\frac{1}{2}} + R_{j+\frac{1}{2}})^{1/2} \right]^{\alpha-1} (\Delta R)_i.$$

B. The Conservation Equations for Time Advancement

1. Subroutine VELOC (NMIN,NMAX)

$$U_{j+\frac{1}{2}}^{n+1} = U_{j+\frac{1}{2}}^{n-\frac{1}{2}} + \frac{(\Delta t) (R_{j+\frac{1}{2}}^n)^{\alpha-1}}{0.5(M_j + M_{j+1})} \left[(P_j^n - P_{j+1}^n) + (q_j^n - q_{j+1}^n) \right] - \frac{(\alpha-1) (\varphi) (v_j^n + v_{j+1}^n) (\Delta t)}{R_{j+\frac{1}{2}}^n},$$

where

$$\varphi = \frac{3 (Sx_j^n + Sx_{j+1}^n)}{4} \quad \text{for } \alpha = 3,$$

and

$$\varphi = \frac{1}{2} \left[2(Sx_j^n + Sx_{j+1}^n) + (Sz_j^n + Sz_{j+1}^n) \right]$$

for $\alpha = 2$.

2. Subroutine RADIUS (NMIN,NMAX)

$$R_{j+\frac{1}{2}}^{n+1} = R_{j+\frac{1}{2}}^n + U_{j+\frac{1}{2}}^{n+\frac{1}{2}} (\Delta t). \quad (1)$$

3. Subroutine VOLUM (NMIN,NMAX)

$$V_j^{n+1} = \left(\frac{R_{j-\frac{1}{2}}^{n+1} + R_{j+\frac{1}{2}}^{n+1}}{2} \right)^{\alpha-1} \frac{(R_{j+\frac{1}{2}}^{n+1} - R_{j-\frac{1}{2}}^{n+1})}{M_j}.$$

4. Subroutine ENERGY (NMIN,NMAX)

$$I_j^{n+1} = I_j^n + \frac{(\Delta t)}{M_j} \left\{ \left[\frac{M_j P_{j-1}^n + M_{j-1} P_j^n}{M_j + M_{j-1}} + 0.5(q_j^n + q_{j-1}^n) \right] U_{j-\frac{1}{2}}^{n+\frac{1}{2}} (R_{j-\frac{1}{2}}^{n+1})^{\alpha-1} - \left[\frac{M_{j+1} P_j^n + M_j P_{j+1}^n}{M_j + M_{j+1}} + 0.5(q_j^n + q_{j+1}^n) \right] U_{j+\frac{1}{2}}^{n+\frac{1}{2}} (R_{j+\frac{1}{2}}^{n+1})^{\alpha-1} \right\} + \frac{1}{8} \left[(U_{j+\frac{1}{2}}^{n-\frac{1}{2}} + U_{j-\frac{1}{2}}^{n-\frac{1}{2}})^2 - (U_{j+\frac{1}{2}}^{n+\frac{1}{2}} + U_{j-\frac{1}{2}}^{n+\frac{1}{2}})^2 \right].$$

5. Subroutine HETCON (NMIN,NMAX)

$$I_j^{n+1} = I_j^{n+1} + \frac{\lambda(\Delta t)}{M_j} \left[\frac{(R_{j+\frac{1}{2}}^{n+1})^{\alpha-1} (T_{j+1}^n - T_j^n)}{\frac{1}{2}(R_{j+\frac{1}{2}}^{n+1} - R_{j-\frac{1}{2}}^{n+1})} - \frac{(R_{j-\frac{1}{2}}^{n+1})^{\alpha-1} (T_j^n - T_{j-1}^n)}{\frac{1}{2}(R_{j+\frac{1}{2}}^{n+1} - R_{j-\frac{1}{2}}^{n+1})} \right].$$

C. Viscosity - Subroutine VISCOS (NMIN,NMAX)

1. PIC Form

$$q_j^{n+1} = \frac{K}{v_j^{n+1}} (0.5) (U_{j-\frac{1}{2}}^{n+\frac{1}{2}} + U_{j+\frac{1}{2}}^{n+\frac{1}{2}}) (U_{j-\frac{1}{2}}^{n+\frac{1}{2}} - U_{j+\frac{1}{2}}^{n+\frac{1}{2}}),$$

if $(U_{j-\frac{1}{2}}^{n+\frac{1}{2}} - U_{j+\frac{1}{2}}^{n+\frac{1}{2}})$ is positive; otherwise, $q_j^{n+1} = 0$. The absolute value of q is used.

2. Landshoff Form

$$q_j^{n+1} = \frac{K}{v_j^{n+1}} (U_{j-\frac{1}{2}}^{n+\frac{1}{2}} - U_{j+\frac{1}{2}}^{n+\frac{1}{2}}).$$

Restrictive conditions are the same as for the PIC form.

3. "Real" Form

$$q_j^{n+1} = 1.333 \frac{K}{v_j^{n+1}} \frac{(U_{j-\frac{1}{2}}^{n+\frac{1}{2}} - U_{j+\frac{1}{2}}^{n+\frac{1}{2}})}{M_j},$$

where K is the "coefficient of viscosity," and this form is appropriate only for slabs.

D. Burn Techniques - Subroutine BURN (NMIN,NMAX)

for 1 and 2

1. Arrhenius Burn

$$W_j^{n+1} = W_j^n - \Delta t Z W_j^n e^{-E^*/R_g T_j^n}$$

where $1 \geq W \geq 0$.

2. C-J Volume Burn

Assumes that W varies linearly with V from V_o to V_{CJ} .

$$W_j^{n+1} = 1 - \frac{V_o - V_j^{n+1}}{V_o - V_{CJ}}$$

where

$$1 \geq W_j^{n+1} \geq 0,$$

and

$$W_j^n \geq W_j^{n+1}.$$

$$P_j^{n+1} = (1 - W_j^{n+1}) (P^1)$$

where P^1 is pressure of detonation products at V, I, and $W = 0$, if $W_j^{n+1} < 0.99$ (WMAX); otherwise, $P_j^{n+1} = P_o$. The P_j^{n+1} calculation is actually performed in the subroutine EQST.

3. Gamma-Law Taylor Wave Burn

The explosive is burned before the first time interval by assuming it to be a gamma-law explosive that has been detonated with a rear boundary of constant velocity. Coded as part of initial setup.

Knowing ρ_o , γ , D_{CJ} , we compute

$$U_{CJ} = \frac{D_{CJ}}{\gamma + 1}$$

$$P_{CJ} = \frac{\rho_o D_{CJ}^2}{\gamma + 1}$$

$$V_{CJ} = \left(\frac{\gamma}{\gamma + 1}\right) \left(\frac{1}{\rho_o}\right)$$

$$C_{CJ} = D_{CJ} - U_{CJ}$$

$$k = \frac{\gamma - 1}{\gamma + 1}$$

$$l = U_{CJ} - \frac{2}{\gamma - 1} C_{CJ}$$

and

$$t = \frac{R_n - R_o}{D_{CJ}}$$

where $R_n - R_o$ is thickness of explosive.

For each cell with a radius of $R_{j-\frac{1}{2}}$, we compute

$$Y = (R_n - R_{j-\frac{1}{2}})/t$$

$$U_{j+\frac{1}{2}} = \frac{2}{\gamma + 1} Y + kt$$

$$C_j = \frac{Y - l}{2} (U_{j+\frac{1}{2}} - t)$$

$$P_j = P_{CJ} \left(\frac{C_j}{C_{CJ}}\right)^{2\gamma/\gamma-1}$$

$$V_j = \left[\frac{[P_{CJ} (V_{CJ})^\gamma]^{1/\gamma}}{P_j} \right]$$

$$I_j = \frac{P_j V_j}{\gamma - 1} - \frac{P_{CJ} V_{CJ}}{\gamma - 1} + \frac{P_{CJ}}{2} (V_o - V_{CJ})$$

and

$$R_{j+\frac{1}{2}} = R_{j-\frac{1}{2}} + (M_j)(V_j).$$

With the new $R_{j+\frac{1}{2}}$ set equal to $R_{j-\frac{1}{2}}$, we return to compute Y, etc. When $U_{j+\frac{1}{2}}$ is equal to the applied piston velocity, the rest of the explosive cells are made identical to the last cell calculated.

E. The Elastic Stress Deviators - Subroutine STRESS (NMIN,NMAX)

$$Sx_j^{n+1} = Sx_j^n + 2\mu \left[- \left(\frac{U_{j+\frac{1}{2}}^{n+1} - U_{j-\frac{1}{2}}^{n+1}}{R_{j+\frac{1}{2}}^{n+1} - R_{j-\frac{1}{2}}^{n+1}} \right) (\Delta t) + \frac{2}{3} \left(\frac{v_j^{n+1} - v_j^n}{v_j^{n+1} + v_j^n} \right) \right].$$

$$Sz_j^{n+1} = Sz_j^n + \frac{4}{3}\mu \left(\frac{v_j^{n+1} - v_j^n}{v_j^{n+1} + v_j^n} \right).$$

F. The Equation of State and Yield Calculation - Subroutine EQST (NMIN,NMAX)

Enter HOM Subroutine with V_j^{n+1} , I_j^{n+1} , W_j^{n+1} to find P_j^{n+1} .

If $P_j^{n+1} > P_o$,

then

$$(P_{hyd})_j^{n+1} = (P_{hydrostat})_j^{n+1} = P_j^{n+1} - (2/3 Y_o) \left(\frac{P_j^{n+1}}{0.050} \right)$$

where

$$\frac{P_j^{n+1}}{0.05} \leq 1.0.$$

The 0.050 is input PLAP. If $P_j^{n+1} < P_o$, then

$$(P_{hyd})_j^{n+1} = P_j^{n+1}.$$

For $\alpha = 1$ or 3 if

$$|Sx_j^{n+1}| \geq |2/3 Y_o|, P_j^{n+1} = (P_{hyd})_j^{n+1} \pm 2/3 Y_o$$

where the sign of $(2/3 Y_o)$ is identical to the sign of Sx_j^{n+1} . Or if

$$|Sx_j^{n+1}| < |2/3 Y_o|, P_j^{n+1} = (P_{hyd})_j^{n+1} + Sx_j^{n+1}.$$

For $\alpha = 2$

$$f = 2 \left[(Sx_j^{n+1})^2 + Sx_j^{n+1} Sz_j^{n+1} + (Sz_j^{n+1})^2 \right].$$

Then if

$$f < 2/3 (Y_o)^2, P_j^{n+1} = (P_{hyd})_j^{n+1} + Sx_j^{n+1},$$

and

$$Sz_j^{n+1} = Sz_j^{n+1}.$$

Or if

$$f > 2/3 (Y_o)^2, P_j^{n+1} = (P_{hyd})_j^{n+1} + \left(\sqrt{\frac{2/3 (Y_o)^2}{f}} \right) (Sx_j^{n+1}),$$

and

$$Sz_j^{n+1} = (Sz_j^{n+1}) \left(\sqrt{\frac{2/3 (Y_o)^2}{f}} \right).$$

G. Total Energy of Components Coded as Part of Subroutine TIME

$$\text{Kinetic energy} = \sum_1^j \xi M_j \frac{(U_{j+\frac{1}{2}}^{n+1})^2}{(2)(0.9992768)}.$$

$$\text{Internal energy} = \sum_1^j \xi M_j \left(I_j^{n+1} \right),$$

where $\xi = 4\pi$ for spheres, 2π for cylinders, and 1 for slabs; and the units of energy are Mbar-cc/g.

H. The Boundary Conditions

1. An Applied Piston on Right Boundary

$U_{j+\frac{1}{2}}^n = U_{piston}$ where outside boundary is $j+\frac{1}{2}$, if $W_{j-\frac{1}{2}}^n > 0.5$; otherwise = final U_{piston} , or

$U_{piston} = A + (B)(\text{Time})$. For both,

$$P_{j+1}^n = P_j^n, \text{ and } Q_{j+1}^n = Q_j^n.$$

2. An Applied Piston on Left Boundary

$U_{\frac{1}{2}}^n = U_{\frac{1}{2}}^{n+1} = U_{piston}$ where boundary is at $j = \frac{1}{2}$ if $W_{j+\frac{1}{2}}^n > 0.5$; otherwise = final U_{piston} , or

$U_{piston} = A + (B)(\text{Time})$. For both,

$$P_o^n = P_1^n, \text{ and } q_o^n = q_1^n.$$

$R_{\frac{1}{2}}$ is computed by Equation (1).

3. A Steady-State Reaction Zone Piston on Left Boundary

The steady-state reaction zone piston is computed by iteration for a given detonation velocity by using the amount of reaction that occurred in a cell near the piston to determine the proper particle velocity of the piston.

For a W , iterate on V using linear feedback by calculating

$$I = I_o + \frac{1}{2}(\rho_o)^2 (D_{CJ})^2 (V_o - V)^2$$

and

$$P_r = P_o + (\rho_o)^2 (D_{CJ})^2 (V_o - V)$$

where P_r is Rayleigh line pressure.

Using the HOM equation of state, calculate P for V , I , and W . Continue iteration until $P_r - P \leq 1 \times 10^{-5}$. Calculate $U = \sqrt{(P - P_o)(V_o - V)}$ and assign it to final U_{piston} .

4. A Right Free-Surface Boundary

$$P_{j+1}^n = -P_j^n \text{ where outside boundary is } j+\frac{1}{2}.$$

$$U_{j+1}^n = U_j^n.$$

5. A Left Free-Surface Boundary

$$P_0^n = -P_1^n \text{ where boundary is at } j = \frac{1}{2}.$$

$$U_{\frac{1}{2}}^n = -U_{1\frac{1}{2}}^n.$$

$$q_0^n = q_1^n.$$

6. A Right Continuum Boundary

$$P_{j+1}^n = P_j^n \text{ where outside boundary is } j+\frac{1}{2}.$$

7. A Left Continuum Boundary

$$P_0^n = P_1^n \text{ where boundary is at } j = \frac{1}{2}.$$

$$q_0^n = q_1^n.$$

$$U_{\frac{1}{2}}^n = U_{1\frac{1}{2}}^n \text{ unless } \alpha = 2 \text{ or } 3 \text{ when } U_{\frac{1}{2}}^n = -U_{1\frac{1}{2}}^n.$$

III. THE HOM EQUATION OF STATE SUBROUTINE

HOM is a FORTRAN subroutine which calculates the pressure and temperature given the internal energy, specific volume, and mass fraction of the solid for solids, gases, and mixtures.

The Nomenclature

C,S	coefficients to a linear fit of U_s and U_p
Cl,S1	second set of coefficients to a linear fit of U_s and U_p
C_V	heat capacity of condensed component (cal/g/deg)
C'_V	heat capacity of gaseous component (cal/g/deg)
I	total internal energy (Mbar-cc/g)
P	pressure (Mbar)
SPA	spalling constant to relate spall pressure and tension rate
SPALL P	interface spalling pressure
T	temperature ($^{\circ}$ K)
USP	ultimate spalling pressure
U_p	particle velocity
U_s	shock velocity
V	total volume (cc/g)
V_0	initial volume of condensed component (cc/g)

W mass fraction of undecomposed explosive

Subscripts

g	gaseous component
H	Hugoniot
i	isentropie
s	condensed component

The Method

A. Condensed Components

(The mass fraction, W, is 1; the internal energy, I, is I_s ; and the specific volume, V, is V_s). For volumes less than V_0 , the experimental Hugoniot data are expressed as a linear fit of the shock and particle velocities. The Hugoniot temperatures are computed using the code described in Appendix C.

$$U_s = C + SU_p.$$

$$P_H = \frac{C^2(V_0 - V_s)}{[V_0 - S(V_0 - V_s)]^2}.$$

$$\ln T_H = F_s + G_s \ln V_s + H_s (\ln V_s)^2 + I_s (\ln V_s)^3 + J_s (\ln V_s)^4.$$

$$I_H = \frac{1}{2} P_H (V_0 - V_s).$$

$$P_s = \frac{V_s}{V_0} (I_s - I_H) + P_H, \text{ where } \gamma_s = v \left(\frac{\partial P}{\partial E} \right)_V. \quad (2)$$

$$T_s = T_H + \frac{(I_s - I_H)(23,890)}{C_V}. \quad (3)$$

Two sets of C and S coefficients may be given. For $V_s < \text{MINV}$, the fit $U_s = C1 + S1(U_p)$ is used with the corresponding changes to the above equations. Between MINV and VSW, the volume is set equal to MINV, and $U_s = C1 + S1(U_p)$ is used. For volumes greater than V_0 , we use the Grüneisen equation of state and the $P = 0$ line as the standard curve.

$$P_s = \left[I_s - \frac{C_V}{(3)(23890)(\alpha)} \left(\frac{V_s}{V_0} - 1 \right) \right] \frac{V_s}{V_0}.$$

$$T_s = \frac{(I_s)(23,890)}{C_V} + T_0.$$

The spalling option is not used if $SPA < 0.0001$. If $P_s \leq USP$, set $P_s = \text{SPALL P}$ and set spall indicator. If $P_s \leq SPA \sqrt{\Delta P / \Delta X}$ ($\Delta P / \Delta X$ is the tension

rate), and $P_s \leq \text{SPMIN}$ (5×10^{-3}), set $P_s = \text{SPALL P}$ and set spall indicator. Do not spall if neither of the above conditions are satisfied.

B. Gas Components

(Mass fraction, W , is 0; the internal energy, I , is I_g ; and the specific volume, V , is V_g). The pressure, volume, temperature, and energy values of the detonation products are computed using FORTRAN BKW⁸ and fitted by a method of least squares to Eq. (4) through (6). A gamma-law gas may also be fit to these equations as a special case. A code to perform this is described in Appendix C.

$$\ln P_i = A + B \ln V_g + C (\ln V_g)^2 + D (\ln V_g)^3 + E (\ln V_g)^4. \quad (4)$$

$$\ln I_i = K + L \ln P_i + M (\ln P_i)^2 + N (\ln P_i)^3 + O (\ln P_i)^4. \quad (5)$$

$I_i = I_i - Z$ (where Z is a constant used to change the standard state to be consistent with the solid explosive standard state, and if the states are the same is used to keep I positive when making a fit).

$$\ln T_i = Q + R \ln V_g + S (\ln V_g)^2 + T (\ln V_g)^3 + U (\ln V_g)^4. \quad (6)$$

$$-\frac{1}{3} = R + 2S \ln V_g + 3T (\ln V_g)^2 + 4U (\ln V_g)^3.$$

$$P = \left(\frac{1}{9V_i} \right) (I_g - I_i) + P_i. \quad (7)$$

$$T = T_i + \frac{(I_g - I_i)(23,890)}{C_V} \quad (8)$$

C. Mixture of Condensed and Gaseous Components

($0 < W < 1$)

$$V = WV_g + (1 - W)V_s.$$

$$I = WI_s + (1 - W)I_g.$$

$$P = P_g = P_s.$$

$$T = T_g = T_s.$$

Multiplying Eq. (3) by (W/C_V) and Eq. (8) by $(1 - W)/C_V$ and adding, we get, after substituting T for T_s and T_g and I for $WI_s + (1 - W)I_g$,

$$T = \frac{23,890}{C_V W + C_V'(1 - W)} \left\{ I - [WI_H - I_i(1 - W)] + \frac{1}{23,890} [T_H C_V W + T_i C_V'(1 - W)] \right\}. \quad (9)$$

Equating Eq. (2) and (7) and substituting from (9), we get

$$P_H - P_i + \left(\frac{Y_s C_V}{V_s} - \frac{C_V'}{8V_g} \right) \left(\frac{1}{C_V W + C_V'(1 - W)} \left\{ I - [WI_H + I_i(1 - W)] + \frac{1}{23,890} [T_H C_V W + T_i C_V'(1 - W)] \right\} \right) - \frac{1}{23,890} \left(\frac{Y_s C_V T_H}{V_s} - \frac{C_V' T_i}{8V_g} \right) = 0. \quad (10)$$

Knowing V , I , and W , one may use the linear feedback to iterate on either V_s or V_g until Eq. (10) is satisfied.

For $V < V_0$, we iterate on V_s with an initial guess of $V_s = V_0$ and a ratio to get the second guess of 0.999. For $V \geq V_0$, we iterate on V_g with an initial guess of $V_g = (V - 0.9 V_0 W)/(1 - W)$ and a ratio to get the second guess of 1.002.

If the iteration goes out of the physical region ($V_g \leq 0$ or $V_s \leq 0$), that point is replaced by $V_s = V_g = V$. Then knowing V_s and V_g , we calculate P and T .

The Calling Sequence

Call HOM (V , S , G , IND)

V , S , and G are dimensioned arrays of size 5, 23, and 17 numbers, respectively.

V(1) specific volume V

V(2) internal energy I

V(3) mass fraction W

V(4) $-\left| \frac{\Delta P}{\Delta x} \right|$ input; pressure P output

V(5) temperature T output

S(1) C

S(2) S

S(3) VSW

S(4) C1

S(5) S1

S(6) F

S(7) G

S(8) H

S(9) I

S(10) J
 S(11) γ_s
 S(12) C_V
 S(13) V_0
 S(14) α
 S(15) SPA
 S(16) USP
 S(17) T_0
 S(18) P_0
 S(22) SPALL P
 S(23) MINV
 G(1) A
 G(2) B
 G(3) C
 G(4) D
 G(5) E
 G(6) K
 G(7) L
 G(8) M
 G(9) N
 G(10) O
 G(11) Q
 G(12) R
 G(13) S
 G(14) T
 G(15) U
 G(16) C'_V
 G(17) Z

Ind set to 0 for normal exit, to -1 for iteration error in mixture calculations, and to +1 for a spalled solid.

APPENDIX A
 FORTRAN SIN INPUT

The following is a description of the input to FORTRAN SIN.

A few rules for setting up a problem are:

1. The steady-state reaction zone piston is suitable only for slab geometry and only at the left boundary.

2. The gamma-law Taylor wave option assumes that the user is using a gamma-law HOM equation of state. It is suitable only for slab geometry and assumes that the explosive has been burned from right to left. The option assumes that the user is using a right-boundary, initial-final velocity piston and

uses the final velocity as the lowest particle velocity permitted in the detonation products Taylor wave.

3. The real viscosity option is suitable only for slab geometry.

4. The spalling option is suitable only for problems with approximately constant tension gradients and many mesh points.

5. The thermal conductivity option assumes that the system will have uniform thermal conductivity.

6. To restart from tape, one needs only to put the last tape dump number in column 1-5 of the first card. This tape dump number is printed on the listings. When the number in column 1-5 of the first card is greater than 10, the code searches through the tape dumps for a dump at a cycle greater than or equal to the number on the input card. When the proper dump is found, all the variables are read and the calculation proceeds.

Col.	Format	1st card
1-5	I5	Number of components or, if > 10, the tape restart cycle number
6-10	I5	Total number of space increments
11-15	I5	Not used
16-20	I5	Not 0 for 4020 output
21-25	I5	Not 0 for graph of pressure vs. radius
26-30	I5	Not 0 for graph of temperature vs. radius
31-35	I5	Not 0 for graph at mass fraction vs. radius
36-40	I5	Not 0 for graph of volume vs. radius
41-45	I5	Not 0 for graph of particle velocity vs. radius
46-50	I5	Not 0 for inclusion of heat conduction
51-55	I5	Left boundary indicator 0 for continuum 1 for free surface 2 for initial velocity-final velocity piston 3 for A + BT velocity piston 4 for steady-state reaction zone piston
56-60	I5	Right boundary indicator

		0 for continuum			<u>Next card</u>
		1 for free surface	<u>Col.</u>	<u>Format</u>	
		2 for initial velocity-final velocity piston	1-18	E18.11	Right boundary card - same as left boundary card.
		3 for A + BF velocity piston	etc.	etc.	Again if not a piston, there is no card. A right steady-state reaction zone piston is not permitted. A right boundary initial-final velocity piston is required if a gamma-law Taylor wave burn is requested.
		<u>2nd card</u>			
<u>Col.</u>	<u>Format</u>				
1-72	12A6	72 columns of alphabetic label for problem			
		<u>3rd card</u>			<u>Next card</u>
<u>Col.</u>	<u>Format</u>		<u>Col.</u>	<u>Format</u>	
1-18	E18.11	Alpha - 1. for slab geometry			If heat conductivity is included, this card contains:
		2. for cylindrical geometry			
		3. for spherical geometry			
19-36	E18.11	Print every this many cycles	1-18	E18.11	Heat conduction constant - same for all components.
37-54	E18.11	Graph every this many cycles			
55-72	E18.11	Tape dump every this many cycles			
		<u>4th card</u>			The following cards are present for each component.
<u>Col.</u>	<u>Format</u>				<u>1st comp. card</u>
		Left boundary card - If a left boundary initial-final piston is requested, this card is:	<u>Col.</u>	<u>Format</u>	
			1-12	2A6	Alphabetic name of component
					<u>2nd comp. card</u>
1-18	E18.11	Initial piston velocity	<u>Col.</u>	<u>Format</u>	
19-36	E18.11	Final piston velocity	1-5	I5	Number of space increments for this component
37-41	I5	Number of cells from left initially in calculation; if 0, all cells are used.	6-10	I5	Not 0 for gas or explosive
			11-15	I5	0 for Arrhenius burn, 1 for C-J volume burn, 2 for gamma-law Taylor wave
42-46	I5	Add on a cell every this many cycles.			
		If a left-boundary A + BF piston is requested, this card is:	16-20	I5	0 for PIC viscosity, 1 for Landshoff viscosity, 2 for real viscosity
1-18	E18.11	A			
19-36	E18.11	B			
37-41	I5	Same as initial-final piston			
42-46	I5	Same as initial-final piston			
		If a left steady-state piston is requested, this card is:	<u>Col.</u>	<u>Format</u>	<u>3rd comp. card</u>
1-18	E18.11	Detonation velocity	1-18	E18.11	Space increment (cm)
19-36	E18.11	Spike volume	19-36	E18.11	Time increment (usec)
37-41	I5	Same as initial-final piston	37-54	E18.11	Viscosity factor
42-46	I5	Same as initial-final piston	55-72	E18.11	Initial density (g/cc)
					<u>4th comp. card</u>
		If a piston is not requested, no card appears for the left boundary.	<u>Col.</u>	<u>Format</u>	
			1-18	E18.11	Initial pressure (Mbar)
			19-36	E18.11	Initial temperature (°K)
			37-54	E18.11	Initial internal energy (Mbar-cc/g)
			55-72	E18.11	Initial velocity (cm/usec)

5th-10th comp. card

Six cards of solid parameters for equation of state and elastic-plastic calculations. The variables are four per card with each card of the format 4E18.11. They are, in order: C, S, VSW, Cl, Sl, F, G, H, I, J, γ_s , C_V , V_0 , α , SPA, USP, T_0 , P_0 , $2/3 Y_0$, μ , PLAP, SPALL P, and MINV.

<u>Col.</u>	<u>Format</u>	<u>11th comp. card</u>
1-18	E18.11	Initial mass fraction of the solid
19-36	E18.11	Frequency factor for Arrhenius burn
37-54	E18.11	Activation energy for Arrhenius burn
55-72	E18.11	Volume for C-J burn or C-J detonation velocity for a γ -law Taylor wave

For a pure solid there are no more cards.

For a gas or explosive we have,

12th-16th comp. cards

Five cards of gas equation of state parameters. The variables are four to a card with each card of the format 4E18.11. They are, in order: A, B, C, D, E, K, L, M, N, O, Q, R, S, T, U, C'_V , and Z.

APPENDIX B

THE FORTRAN SIN CODE

In this appendix we list the names or symbols of the variables in SIN both as used in the FORTRAN code and in this report. The listing of the FORTRAN code as it existed at the time of preparation of this report is also presented. Errors in this version of the code will be corrected as they become known to the authors. Anyone wishing to actually copy any part of this code should contact the authors for the latest version.

Variables Used in FORTRAN SIN

<u>Code Name</u>	<u>Report Symbol</u>
CR - cell radius	$R_{j-\frac{1}{2}}^{n+1}$
CRO - previous cell radius	$R_{j-\frac{1}{2}}^n$
CU - cell velocity	$U_{j-\frac{1}{2}}^{n+\frac{1}{2}}$
CUO - previous cell velocity	$U_{j-\frac{1}{2}}^{n-\frac{1}{2}}$

CV - cell volume	V_j^{n+1}
CVO - previous cell volume	V_j^n
CM - cell mass	M_j
CQ - cell viscosity	Q_j^n
ICF - cell flag word	

XXYY₁ - YY component index

- XX spall flag

0 - no spall

1 - spalled cell

2 - interface cell

CW - cell mass fraction	W_j^{n+1}
CWO - previous cell mass fraction	W_j^n
CT - cell temperature	T_j^n
CP - cell pressure	P_j^n
CS - cell stress deviator in x or r direction	Sx_j^n

CSZ - cell stress deviator in z direction	Sz_j^n
CI - cell internal energy	I_j^{n+1}
CIO - previous cell internal energy	I_j^n

LABEL - 12 words containing 72 columns of alphabetic information - the problem label - format 12A6

TTIME - total time elapsed in problem
DELT - time increment for this cycle
ICYCL - cycle number
PCNT - count for printing
GCNT - count for graphing
DCNT - count for dumping
PINC - print every PINC cycles
GINC - graph every GINC cycles
DINC - dump every DINC cycles
IALPH - 0 for slab, 1 for cylinder, 2 for sphere
HEATC - heat conductivity constant
GASW - set W to 0 if less than GASW
NCL - total number of cells in calculation
PTMIN - the pressure of one cell of a component must be greater than PTMIN ($1. \times 10^{-5}$) to use that component's delta t.
IPNT - unused

IMC - not 0 for microfilm output
 IPR - not 0 for a pressure vs. radius graph
 ITR - not 0 for a temperature vs. radius graph
 IWR - not 0 for a mass fraction vs. radius graph
 IIR - not 0 for an internal energy vs. radius graph
 IUR - not 0 for a particle velocity vs. radius graph
 IHC - not 0 to include heat conduction
 ILB - left boundary indicator

<u>At Read-In</u>	<u>During Execution</u>	
0	1	free surface
1	2	continuum
2	3	initial-final piston
3	4	A + BT piston
4	5	steady-state reaction zone piston

IRB - right boundary indicator (same as left except no reaction zone piston)
 WSW - W to switch from initial to final for that piston
 WMAX - for $W > WMAX$ (0.99) in C-J burn, $P = P_0$
 SMIN - change in stress set to 0 if less than SMIN (1.0×10^{-9})
 ALB, BLB - left boundary piston constants: either initial and final velocities or A and B in A + BT
 ARB, BRB - right boundary piston constants (same as left)
 NLI - number of cells initially in calculation from left. If 0, all cells are used.
 NLINC - add on a cell from left every NLINC cycles
 NLCNT - count for adding on cells from left
 NRI, NRINC, NRCNT - same as above, only from right. Note: These numbers are used only if there is a piston at their respective boundaries. If there is no piston specified, all cells are used.
 NLH - number of cells from the left in calculation at the time of interest
 NRL - index of lowest cell in calculation from right end

DX - at input-space increment for each component during calculation - space increment at right-most component
 E - activation energy for Arrhenius burn
 GAS - gas parameters
 SOL - solid parameters
 VCJ - C-J burn volume
 VFACT - viscosity factor
 Z - frequency factor for Arrhenius burn
 NAM - alphabetic names of components (2 words per component)
 DTIME - delta t for each component
 NINC - number of space increments for each component
 NCOM - number of components
 IEXP - not 0 for explosive or gas
 IERN - 0 for Arrhenius burn, 1 for C-J volume burn, 2 for gamma-law Taylor wave
 IVIS - 0 for PIC viscosity, 1 for Landshoff form, 2 for real form
 ALPH - 1 for slab, 2 for cylinder, 3 for sphere
 RHO0 - initial density, only for the component being read
 P0 - initial pressure, only for the component being read
 T0 - initial temperature, only for the component being read
 E0 - initial internal energy, only for the component being read
 U0 - initial particle velocity, only for the component being read
 W0 - initial mass fraction, only for the component being read
 NINC - number of space increments for this component
 RFRNT - r coordinate value of the nearest spalled or interface cell of the same component with a lower index
 EQ - five parameters used to call HOM. EQ(1) - volume, EQ(2) - energy, EQ(3) - mass fraction, EQ(4) - $|\Delta P/\Delta X|$ for spall input or pressure output, EQ(5) - temperature output
 INDH - HOM indicator -1 error, 0 normal, +1 spalled
 X - elastic-plastic ratio of P/PLAP, not greater than 1
 DELS - change in stress deviator
 TKE - total kinetic energy of each component

TIE - total internal energy of each component
LIN - line counter for printing
KIND - the index of cell on right-hand boundary
DCJ - C-J detonation velocity for γ -law Taylor
wave burn
DRT - total delta R for γ -law Taylor wave burn
DRF - first R for γ -law Taylor wave burn
DRL - last R for γ -law Taylor wave burn
UCJ, PCJ, VCR, CCJ - C-J values for γ -law
Taylor wave burn
COK - $(\gamma - 1)/(\gamma + 1)$ for γ -law Taylor wave
burn
COL - $UCJ - \frac{2C}{\gamma - 1}$ for γ -law Taylor wave burn
RXZDV - steady-state reaction zone piston detona-
tion velocity
RXZVS - steady-state reaction zone piston spike
volume guess

Listing of the FORTRAN Code

```

PROGRAM FSIN (INPUT,OUTPUT,FILM,TAPE12=FILM,TAPE1)
C   CR      CELL RADIUS  J-1/2
C   CU      CELL VELOCITY J-1/2
C   CV      CELL VOLUME  J
C   CM      CELL MASS    J
C   CQ      CELL VISCOSITY J
C   ICF     CELL FLAG WORD J
C   CW      CELL MASS FRACTION J
C   CT      CELL TEMPERATURE J
C   CP      CELL PRESSURE J
C   CS      CELL STRESS DEVIATOR J
C   CSZ     CELL STRESS DEVIATOR IN Z DIRECTION J
C   CI      CELL INTERNAL ENERGY J
000003  COMMON CR(999),CRO(999),CU(999),CUO(999),CV(999),CVO(999),
ICM(999),CQ(999),ICF(999),CW(999),CWO(999),CT(999),CP(999),
2CS(999),CSZ(999),CI(999),CIO(999),LABEL(12)
000003  COMMON /MISC/ TTIME,DELT,ICYCL,PCNT,GCNT,DCNT,DINC,PINC,GINC,
1IALPH,HEATC,GASW,NCL,PTMIN,IPNT,IMC,IPR,ITR,IWR,IIR,IUR,IHC,ILB,
2IRB,WSW,PSMIN,WMAX,SMIN,ALB,BLB,ARB,BRB,NLI,NRI,NLINC,NRINC,
3NLFNT,NRCNT,NLH,NRL,DX,KIND,RXZDV,RXZVS,FPO,FEO,RXZ
000003  COMMON /MATER/ E(10),GAS(17,10),IBRN(10),SDL(23,10),VCJ(10),
1VFACT(10),Z(10),NAM(2,10),DIME(10),NOINC(10),IVIS(10),IEXP(10),
2NCOM
000003  DIMENSION RXZ(10),EQ(5)
C   WSW     W TO SWITCH PISTON VELOCITY FROM INITIAL TO FINAL
000003  DATA WSW /0.02/
000003  DATA NRI /0/
000003  DATA NLI /0/
000003  DATA NLH /1/
C   COUNTS ARE SET LARGE SO CELLS ARENT ADDED ON IF NOT REQUESTED
000003  DATA NRCNT/100000/
000003  DATA NLCNT/100000/
C   READ INPUT DATA
C   NCOM    NUMBER OF COMPONENTS OR RESTART CYCLE NUMBER
C   NCL     TOTAL NUMBER OF CELLS IN THE CALCULATION
C   IPNT    UNUSED
C   IMC     NOT 0 FOR MICROFILM OUTPUT
C   IPR     NOT 0 TO GIVE A GRAPH OF PRESSURE
C   ITR     NOT 0 TO GIVE A GRAPH OF TEMPERATURE
C   IWR     NOT 0 TO GIVE A GRAPH OF MASS FRACTION
C   IIR     NOT 0 TO GIVE A GRAPH OF VOLUME
C   IUR     NOT 0 TO GIVE A GRAPH OF PARTICLE VELOCITY
C   IHC     NOT 0 FOR CALCULATION OF HEAT CONDUCTION IN ENERGY
C   ILB     LEFT HAND BOUNDARY INDICATOR
C           = 0 FOR CONTINUUM
C           = 1 FOR FREE SURFACE
C           = 2 FOR INITIAL-FINAL PISTON
C           = 3 FOR A+BT PISTON
C           = 4 FOR STEADY STATE REACTION ZONE SLAB PISTON
C   IRB     RIGHT HAND BOUNDARY INDICATOR
C           = 0 FOR CONTINUUM
C           = 1 FOR FREE SURFACE
C           = 2 FOR INITIAL-FINAL PISTON
C           = 3 FOR A+BT PISTON
000003  READ 901,NCOM,NCL,IPNT,IMC,IPR,ITR,IWR,IIR,IUR,IHC,ILB,IRB
C   SEE IF TAPE RESTART
000037  IF (NCOM.GT.10) GO TO 500
C   LABEL   72 COLUMNS OF ALPHABETIC COMMENT
000043  READ 900,LABEL
C   ALPH   = 1. FOR SLAB
C           = 2. FOR CYLINDER

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C           = 3. FOR SPHERE
C           PINC   PRINT EVERY PINC CYCLES
C           GINC   GRAPH EVERY GINC CYCLES
C           DINC   DUMP EVERY DINC CYCLES
000050     READ 902,ALPH,PINC,GINC,DINC
000064     IALPH=ALPH-1.
C           READ IN PISTON QUANTITIES
C           ALB,BLB PISTON CONSTANTS FOR LEFT PISTON
C           NLI    NUMBER OF CELLS INITIALLY IN CALCULATION FROM LEFT
C           NLINC  ADD ON A CELL FROM LEFT EVERY NLINC CYCLES
C           RXZDV  STEADY STATE REACTION ZONE PISTON DETONATION VELOCITY
C           RXZVS  STEADY STATE REACTION ZONE PISTON GUESSED SPIKE VOLUME
000067     IF (ILB.EQ.2.OR.ILB.EQ.3) READ 903,ALB,BLB,NLI,NLINC
000111     IF (ILB.EQ.4) READ 903,RXZDV,RXZVS,NLI,NLINC
C           NRI    NUMBER OF CELLS INITIALLY IN CALCULATION FROM RIGHT
C           NRINC  ADD ON A CELL FROM RIGHT EVERY NRINC CYCLES
C           ARB,BRB PISTON CONSTANTS FOR RIGHT PISTON
000127     IF (IRB.EQ.2.OR.IRB.EQ.3) READ 903,ARB,BRB,NRI,NRINC
C           READ IN HEAT CONSTANT
000152     IF (IHC.NE.0) READ 902,HEATC
000161     K=2
000162     CRO(2)=0.
000163     CR(2)=0.
C           IALPH = 0 FOR SLAB
C           1 FOR CYLINDER
C           2 FOR SPHERE
000164     IF (IALPH.EQ.0) PRINT 904,LABEL
000172     IF (IALPH.EQ.1) PRINT 905,LABEL
000202     IF (IALPH.EQ.2) PRINT 906,LABEL
000212     PRINT 907,NCOM
000220     PRINT 908,NCL
000226     IF (ILB.EQ.0) PRINT 909
000233     IF (ILB.EQ.1) PRINT 910
000241     IF (ILB.EQ.2) PRINT 911,ALB,BLB,WSW,NLI,NLINC
000261     IF (ILB.EQ.3) PRINT 912,ALB,BLB,NLI,NLINC
000277     IF (ILB.EQ.4) PRINT 930,RXZDV,RXZVS,NLI,NLINC
000315     IF (IRB.EQ.0) PRINT 913
000322     IF (IRB.EQ.1) PRINT 914
000330     IF (IRB.EQ.2) PRINT 915,ARB,BRB,WSW,NRI,NRINC
000350     IF (IRB.EQ.3) PRINT 916,ARB,BRB,NRI,NRINC
000366     PRINT 917
C           DO 4020 PRINTING IF REQUESTED
000372     IF (IMC.EQ.0) GO TO 1
C           *****
C           4020 OUTPUT BELOW
C           *****
000373     IF (IALPH.EQ.0) WRITE (12,904) LABEL
000402     IF (IALPH.EQ.1) WRITE (12,905) LABEL
000412     IF (IALPH.EQ.2) WRITE (12,906) LABEL
000422     WRITE (12,907) NCOM
000430     WRITE (12,908) NCL
000436     IF (ILB.EQ.0) WRITE (12,909)
000443     IF (ILB.EQ.1) WRITE (12,910)
000451     IF (ILB.EQ.2) WRITE (12,911) ALB,BLB,WSW,NLI,NLINC
000471     IF (ILB.EQ.3) WRITE (12,912) ALB,BLB,NLI,NLINC
000507     IF (ILB.EQ.4) WRITE (12,930) RXZDV,RXZVS,NLI,NLINC
000525     IF (IRB.EQ.0) WRITE (12,913)
000532     IF (IRB.EQ.1) WRITE (12,914)
000540     IF (IRB.EQ.2) WRITE (12,915) ARB,BRB,WSW,NRI,NRINC
000560     IF (IRB.EQ.3) WRITE (12,916) ARB,BRB,NRI,NRINC
000576     WRITE (12,917)
C           *****

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```

C          4020 OUTPUT ABOVE
C          *****
000602    C      READ DATA AND SET UP MESH FOR EACH COMPONENT
          C      1 DO 10 I=1,NCOM
000604    C      NAM      NAME OF COMPONENT
          C      READ 900,NAM(1,I),NAM(2,I)
          C      NINC     NUMBER OF SPACE INCREMENTS FOR THIS COMPONENT
          C      IEXP     NOT 0 FOR GAS OR EXPLOSIVE
          C      IBRN     = 0 FOR ARREHNIUS BURN
          C              = 1 FOR CJ VOLUME BURN
          C              = 2 FOR GAMMA LAW TAYLOR WAVE
          C      IVIS     = 0 FOR PIC VISCOSITY
          C              = 1 FOR LANDSHOFF VISCOSITY
          C              = 2 FOR REAL VISCOSITY
000617    C      READ 901,NINC,IEXP(I),IBRN(I),IVIS(I)
          C      DX      SPACE INCREMENT IN CENTIMETERS
          C      DTIME   TIME INCREMENT IN MICROSECONDS
          C      VFACT   VISCOSITY FACTOR
          C      RH00    INITIAL DENSITY
000633    C      READ 902,DX,DTIME(I),VFACT(I),RH00
          C      P0     INITIAL PRESSURE
          C      T0     INITIAL TEMPERATURE
          C      E0     INITIAL INTERNAL ENERGY
          C      U0     INITIAL VELOCITY
          C      SOL(J,I) SOLID EQUATION OF STATE AND ELASTIC-PLASTIC CONSTANTS
000647    C      READ 902,P0,T0,E0,U0
000663    C      IF(I.GT.1) GO TO 11
000667    C      FPO = P0
000670    C      FE0 = E0
000671    C      11 READ 902,(SOL(J,I),J=1,23)
          C      W0     INITIAL MASS FRACTION
          C      Z      FREQUENCY FACTOR FOR ARREHNIUS BURN
          C      E      ACTIVATION ENERGY FOR ARREHNIUS BURN
          C      VCJ    VOLUME FOR CJ BURN OR NET VEL FOR GAMMA LAW TAYLOR WAVE
000705    C      READ 902,W0,Z(I),E(I),VCJ(I)
000721    C      PRINT 918,NAM(1,I),NAM(2,I),NINC,DX,DTIME(I)
000743    C      IF (IEXP(I).EQ.0) GO TO 2
000745    C      READ 902,(GAS(J,I),J=1,17)
000760    C      PRINT 919
000764    C      IF (IBRN(I).EQ.0) PRINT 921,E(I),Z(I)
000775    C      IF (IBRN(I).EQ.1) PRINT 922,VCJ(I),GASW
001007    C      IF (IBRN(I).EQ.2) PRINT 929,VCJ(I)
001017    C      2 IF (IVIS(I).EQ.0) PRINT 923,VFACT(I)
001026    C      IF (IVIS(I).EQ.1) PRINT 924,VFACT(I)
001036    C      IF (IVIS(I).EQ.2) PRINT 925,VFACT(I)
001046    C      PRINT 926,RH00,P0,T0,E0,U0,W0
001066    C      PRINT 927,(SOL(J,I),J=1,23)
001102    C      IF (IEXP(I).NE.0) PRINT 928,(GAS(J,I),J=1,17)
001117    C      IF (IMC.EQ.0) GO TO 4
          C      *****
          C          4020 OUTPUT BELOW
          C      *****
001120    C      WRITE (12,918) NAM(1,I),NAM(2,I),NINC,DX,DTIME(I)
001142    C      IF (IEXP(I).EQ.0) GO TO 3
001144    C      WRITE (12,919)
001147    C      IF (IBRN(I).EQ.0) WRITE (12,921) E(I),Z(I)
001160    C      IF (IBRN(I).EQ.1) WRITE (12,922) VCJ(I),GASW
001172    C      IF (IBRN(I).EQ.2) WRITE (12,929) VCJ(I)
001202    C      3 IF (IVIS(I).EQ.0) WRITE (12,923) VFACT(I)
001211    C      IF (IVIS(I).EQ.1) WRITE (12,924) VFACT(I)
001221    C      IF (IVIS(I).EQ.2) WRITE (12,925) VFACT(I)
001231    C      WRITE (12,926) RH00,P0,T0,E0,U0,W0

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001251      WRITE (12,927) (SOL(J,I),J=1,23)
001265      IF (IEXP(I).NE.0) WRITE (12,928) (GAS(J,I),J=1,17)
C          *****
C          4020 OUTPUT ABOVE
C          *****
C          FILL IN CELL VALUES FOR THIS COMPONENT
001302      DO 9 J=1,NINC
001304      4 IF (IBRN(I).EQ.2) GO TO 12
001307      CRO(K+1)=CRO(K)+DX
001312      CR(K+1)=CRO(K+1)
001313      CUO(K)=UO
001315      CU(K)=CUO(K)
001316      CP(K)=PO
001320      CT(K)=T0
001321      CW0(K)=W0
001323      CW(K)=CW0(K)
001324      CSZ(K) = 0.
001325      CS(K) = 0.
001326      CIO(K)=E0
001330      CI(K)=CIO(K)
001331      CM(K)=RH00*DX*((CRO(K)+CRO(K+1))*0.5)*IALPH
001342      CVO(K)=1./RH00
001344      CV(K)=CVO(K)
001345      ICF(K)=I
001347      CQ(K)=0.
001350      K=K+1
001351      9 CONTINUE
C          SET SPALL FLAG FOR INTERFACE BETWEEN COMPONENTS
001354      17 IF (IEXP(I).NE.0) GO TO 18
001356      ICF(K-NINC)=ICF(K-NINC)+128
001361      18 NOINC(I) = NINC
001363      10 CONTINUE
C          SET RIGHT AND LEFT BOUNDARY VALUES
001366      CM(1)=CM(2)
001367      CM(K)=RH00*DX*(CRO(K)+DX*0.5)*IALPH
001400      CQ(K)=0.
001401      KIND=K
001402      CRO(1)=-CRO(3)
001404      CRO(K+2)=CRO(K+1)+DX
001406      CI(1)=0.
001406      CS(1)=0.
001407      CSZ(1)=0.
001410      CV(1)=0.
001411      CV(K)=0.
001412      CS(K)=0.
001413      CSZ(K)=0.
001414      CI(K)=0.
001415      ILB=ILB+1
001416      IRB=IRB+1
001417      TTIME=0.
001420      DELT=0.
C          FIND MAXIMUM DELTA T FOR FIRST CYCLE
001421      DO 20 I=1,NCOM
001422      IF (DELT.LT.OTIME(I)) DELT=OTIME(I)
001426      20 CONTINUE
C          SET UP NUMBER OF CELLS IN CALCULATION
001431      NRL=NCL+2
C          SEE IF THERE IS A LEFT PISTON
001433      IF(ILB.LT.2) GO TO 24
C          GO TO 22 IF CELL ADDING ON NOT REQUESTED
001434      IF (NLI.EQ.0) GO TO 22
C          SET NUMBER OF CELLS FROM LEFT AND COUNT FOR ADDING ON CELLS

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001435      NLH=NLI+1
001437      NLCNT=NLINC
001440      C      SEE IF THERE IS A RIGHT PISTON
              IF(IRB.LT.2) GO TO 90
001442      C      ARE CELLS TO BE ADDED ON FROM RIGHT
              IF (NRI.EQ.0) GO TO 90
001443      C      SET INITIAL NUMBER OF CELLS FROM RIGHT AND COUNT FOR ADDING ON
001446      21  NRL=NCL+1-NRI
              NRCNT=NRINC
001447      C      CHECK TO SEE IF ALL CELLS IN CALCULATION
              IF (NRL.GT.NLH+1) GO TO 90
001453      C      USE ALL CELLS IN CALCULATION
001455      22  NLH=NCL+1
001456      NLI=1
001457      NRI=0
              GO TO 90
001460      C      SEE IF THERE IS A RIGHT PISTON
001462      24  IF(IRB.LT.2) GO TO 22
001463      C      ARE CELLS TO BE ADDED ON FROM RIGHT
001464      IF (NRI.EQ.0) GO TO 22
              NLH = 1
              GO TO 21
001465      C      SET LEFT BOUNDARY VALUES
              90  GO TO (100,110,120,130,140),ILB
001476      C      LEFT BOUNDARY CONTINUUM
001500      100 CP(1)=CP(2)
001502      CR(1)=CR(2)-CR(3)
001503      CQ(1)=CQ(2)
001505      CU(1)=-CU(2)
001507      IF (IALPH.EQ.0) CU(1)=CU(2)
001511      CU0(1)=CU(1)
001512      CT(1)=CT(2)
              GO TO 190
001513      C      LEFT BOUNDARY FREE SURFACE
001515      110 CP(1)=-CP(2)
001517      CR(1)=CR(2)-CR(3)
001520      CQ(1)=CQ(2)
001522      CU(1)=-CU(2)
001523      CU0(1)=CU(1)
001524      CT(1)=CT(2)
              GO TO 190
001525      C      LEFT BOUNDARY PISTON - INITIAL,FINAL
001527      120 CQ(1)=CQ(2)
001531      CR(1)=CR(2)-CR(3)
001532      CP(1)=CP(2)
001533      CT(1)=CT(2)
001534      IF (CW(4).LT.WSW) GO TO 125
001536      CU(1)=ALB
001537      CU0(1)=CU(1)
001540      CU(2)=CU(1)
001541      CU0(2)=CU0(1)
001542      GO TO 190
001543      125 CU(1)=RLB
001545      CU0(1)=CU(1)
001546      CU(2)=CU(1)
001547      CU0(2)=CU0(1)
001550      GO TO 190
001551      C      LEFT BOUNDARY PISTON - A*BT
001553      130 CQ(1)=CQ(2)
001555      CR(1)=CR(2)-CR(3)
001556      CP(1)=CP(2)
              CT(1)=CT(I)

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001560      CU(1)=ALB+BLB*TTIME
001563      CU0(1)=CU(1)
001564      CU(2)=CU(1)
001565      CU0(2)=CU0(1)
C          SET RIGHT BOUNDARY VALUES
001567      190 GO TO (200,210,220,230),IRB
C          RIGHT BOUNDARY CONTINUUM
001577      200 CT(K)=CT(K-1)
001601      CP(K)=CP(K-1)
001603      CU(K)=CU(K-1)
001604      CU0(K)=CU0(K-1)
001606      GO TO 290
C          RIGHT BOUNDARY FREE SURFACE
001606      210 CT(K)=CT(K-1)
001610      CP(K)=-CP(K-1)
001612      CU(K)=CU(K-1)
001613      CU0(K)=CU0(K-1)
001615      GO TO 290
C          RIGHT BOUNDARY PISTON - INITIAL,FINAL
001615      220 CT(K)=CT(K-1)
001617      CP(K)=CP(K-1)
001621      IF (CW(K-3).LT.WSW) GO TO 225
001623      CU(K)=ARB
001624      CU0(K)=ARB
001625      GO TO 290
001626      225 CU(K)=BRB
001630      CU0(K)=BRB
001631      GO TO 290
C          RIGHT BOUNDARY PISTON - A*BT
001632      230 CT(K)=CT(K-1)
001634      CP(K)=CP(K-1)
001636      CU(K)=ARB+BRB*TTIME
001641      CU0(K)=CU0(K-1)
001643      290 CONTINUE
001643      IF (NLI.NE.0) CALL VELOC (2,NLH)
001646      IF (NRI.NE.0) CALL VELOC (NRL,NCL+1)
001653      IF (NLI.NE.0) CALL RADIUS (2,NLH+1)
001660      IF (NRI.NE.0) CALL RADIUS (NRL,NCL+2)
C          SET RADIUS FOR CELL BEYOND RIGHT BOUNDARY
001665      CR(K+2)=CR(K+1)*DX
001670      IF (NLI.NE.0) CALL VOLUM (2,NLH)
001673      IF (NRI.NE.0) CALL VOLUM (NRL,NCL+1)
001700      IF (NLI.NE.0) CALL ENERGY (2,NLH)
001703      IF (NRI.NE.0) CALL FNERGY (NRL,NCL+1)
001710      IF (IHC.EQ.0) GO TO 300
001711      IF (NLI.NE.0) CALL HETCON (2,NLH)
001714      IF (NRI.NE.0) CALL HFTCON (NRL,NCL+1)
001721      300 IF (NLI.NE.0) CALL VISCOS (2,NLH)
001724      IF (NRI.NE.0) CALL VISCOS (NRL,NCL+1)
001731      IF (NLI.NE.0) CALL BURN (2,NLH)
001734      IF (NRI.NE.0) CALL BURN (NRL,NCL+1)
C          SET W NEAR AXIS
001741      CW(2)=CW(4)
001743      CW(3)=CW(4)
001744      IF (NLI.NE.0) CALL STRESS (2,NLH)
001747      IF (NRI.NE.0) CALL STRESS (NRL,NCL+1)
001754      IF (NLI.NE.0) CALL EQST (2,NLH)
001757      IF (NRI.NE.0) CALL EQST (NRL,NCL+1)
001764      IF (NLI.NE.0) CALL UPDATE (1,NLH+2)
001771      IF (NRI.NE.0) CALL UPDATE (NRL,NCL+3)
001776      CALL TIME
001777      GO TO 90

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C      RESTART FROM TAPE
002000 500 IRST=NCOM
C      SEARCH FDR DUMP WITH PROPER CYCLE NUMBER
002002 501 READ (1) CRO,CUO,CWO,CSZ,CIO,CVO,ICF,CM,CQ,CT,CP,LABEL,DELT,HEATC,
      1IALPH,TTIME,PINC,GINC,DINC,PCNT,GCNT,OCNT,E,GAS,IBRN,SOL,VCJ,
      2VFACT,Z,NAM,DTIME,NOINC,NCOM,ICYCL,NCL,IPNT,IMC,IPR,ITR,IWR,IIR,
      3IUR,IHC,ILB,IRB,ALB,BLB,ARB,BRB,WSW,NLI,NRI,NLINC,NRINC,NLCNT,
      4NRCNT,NLH,NRL,DX,IVIS,IEXP,KIND,CR,CI,CW,CV,CU,CS,RXZDV,RXZVS,
      SRXZ,FPO,FEO
002227      K=KIND
002231      IF (IRST.GT.ICYCL) GO TO 501
002234      GO TO 90
C      THE SLAB GAMMA LAW TAYLOR WAVE ** ASSUMES DETONATION WAVE MOVED
C      FROM RIGHT TO LEFT AND HAS BRB FINAL PARTICLE VELOCITY
002234 12 L = K
002236      DO 13 J=1,NINC
002237      CR(L+1) = CR(L) + DX
002242      CT(L)=T0
002244      CWO(L)=0.
002245      CW(L)=0.
002246      CSZ(L)=0.
002247      CS(L)=0.
002250      CM(L)=RHO0*DX
002252      CQ(L)=0.
002253      ICF(L)=I
002254      L=L+1
002256 13 CONTINUE
002260      DCJ=VCJ(I)
002261      DRT=CR(L)-CR(K)
002264      DRF=CR(K)
002266      DRL=CR(L)
002267      UCJ=DCJ/(-GAS(2,I)+1.)
002274      PCJ=(RHO0)*(DCJ*DCJ)/(-GAS(2,I)+1.)
002300      VCJR=(-GAS(2,I))/((-GAS(2,I)+1.)*RHO0)
002305      CCJ=DCJ-UCJ
002307      COK=(-GAS(2,I)-1.)/(-GAS(2,I)+1.)
002313      COL=UCJ-(2./(-GAS(2,I)-1.))*CCJ
002320      COT=DRT/DCJ
002322      DO 14 J=1,NINC
002324      COY=(DRL-CR(K))/COT
002330      CU(K)=(2.*COY)/(-GAS(2,I)+1.)+COK*COL
002341      CP(K)=PCJ*(((GAS(2,I)-1.)*0.5)*(CU(K)-COL))/CCJ)**(-GAS(2,I)*2./
      1(-GAS(2,I)-1.))
002361      CV(K)=(PCJ*(VCJR**(-GAS(2,I)))/CP(K))**((1./(-GAS(2,I))))
002376      CVO(K)=CV(K)
002400      CI(K)=(CP(K)*CV(K))/(-GAS(2,I)-1.)-(PCJ*VCJR)/(-GAS(2,I)-1.)
      1+ 0.5*PCJ*(1./RHO0-VCJR)
002417      CIO(K)=CI(K)
002421      DRP=CM(K)*CV(K)
002423      CR(K+1)=CR(K)+DRP
002425      CR0(K+1)=CR(K+1)
002426      CU(K)=-CU(K)
002430      CU0(K)=CU(K)
002431      IF (CU(K).GT.BRB) GO TO 15
002435      K=K+1
002436 14 CONTINUE
002440      GO TO 17
002441 15 L=K
002443      DO 16 N=J,NINC
002445      CU(K)=CU(L)
002450      CU0(K)=CU(L)
002451      CP(K)=CP(L)

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002453      CV(K)=CV(L)
002454      CVO(K)=CV(L)
002455      CI(K)=CI(L)
002456      CIO(K)=CI(L)
002461      CR(K+1)=CR(K)+DRP
002463      CRO(K+1)=CR(K+1)
002465      K=K+1
002466      16 CONTINUE
002470      GO TO 17
C          REACTION ZONE PISTON
C          FOR SLABS AT LEFT BOUNDARY ONLY
002471      DATA RXZ(1)/ 0./
002471      DATA RXZ(2) /1.001/
002471      DATA RXZ(3) /+1.0E-5/
002471      DATA RXZ(10)/ 0./
002471      140 IF(RXZ.EQ.0.) RXZ=RXZVS
002473      IF(CW(4).LT.GASW) GO TO 120
002475      146 CALL LFB (RZV,RZF,RXZ)
002500      IF(RXZ(10)) 141,142,143
002502      141 PRINT 931,RXZ
002510      PCNT=1.
002512      GCNT=100.
002513      ICYCL=ICYCL-1
002515      TTIME=TTIME-DELT
002517      CALL TIME
002520      STOP
002522      931 FORMAT (33H1 REACTION ZONE PISTON LFB ERROR , 5E18.11)
002522      143 RZI=FEO + 0.5/( SOL(13,I)*SOL(13,I))*RXZDV*RXZDV*((SOL(13,I)-
002522      1RZV)**2)
002535      EQ(1)=RZV
002536      EQ(2)=RZI
002537      160 EQ(3)=CW(4)
002541      CALL HOM (EQ,SOL(1,I).GAS(1,I).INDH)
002550      IF(INDH) 144,145,144
002551      144 PRINT 932,I,EQ
002561      932 FORMAT (33H1 HOM LFB ERROR FOR RXZ PISTON ,15.5E18.11)
002561      GO TO 141
002562      145 RXP=1./(SOL(13,I)*SOL(13,I))*RXZDV*RXZDV*(SOL(13,I)-RZV)
002571      RZF=EQ(4)-RXP+ FPO
002574      GO TO 146
002575      142 ALB=SQRT((EQ(4) -FPO)*(SOL(13,I)-RZV))
002606      BLB=ALB
002607      GO TO 120
002607      900 FORMAT (12A6)
002607      901 FORMAT (12I5)
002607      902 FORMAT (4E18.11)
002607      903 FORMAT (2E18.11,2I5)
002607      904 FORMAT (80H1 A SIN ONE DIMENSIONAL REACTIVE HYDRODYNAMIC CALCULAT
002607      1ION IN SLAB GEOMETRY FOR ,/,2X,12A6)
002607      905 FORMAT (87H1 A SIN ONE DIMENSIONAL REACTIVE HYDRODYNAMIC CALCULAT
002607      1ION IN CYLINDRICAL GEOMETRY FOR ,/,2X,12A6)
002607      906 FORMAT (85H1 A SIN ONE DIMENSIONAL REACTIVE HYDRODYNAMIC CALCULAT
002607      1ION IN SPHERICAL GEOMETRY FOR ,/,2X,12A6)
002607      907 FORMAT (//,32H THE NUMBER OF COMPONENTS IS ,I5)
002607      908 FORMAT (//,44H THE TOTAL NUMBER OF SPACE INCREMENTS IS ,I5)
002607      909 FORMAT (//,35H THE LEFT BOUNDARY IS A CONTINUUM)
002607      910 FORMAT (//,38H THE LEFT BOUNDARY IS A FREE SURFACE)
002607      911 FORMAT (//,59H THE LEFT BOUNDARY IS A PISTON WITH INITIAL VELOCIT
002607      1TY OF ,E10.3,24H AND FINAL VELOCITY OF ,E10.3,/,10X,23HWHEN W+3
002607      2IS LESS THAN ,E10.3, 6H WITH ,I5,44H CELLS INITIALLY AND ADDING 0
002607      3N A CELL EVERY ,I5, 7H CYCLES)
002607      912 FORMAT (//,49H THE LEFT BOUNDARY IS A PISTON WITH VELOCITY = .

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1E10.3,1H+.E10.3,6H(TIME),/,10X. 5H WITH ,15,44H CELLS INITIALLY AND
 2 ADDING ON A CELL EVERY ,15, 7H CYCLES)
 002607 913 FORMAT (//,36H THE RIGHT BOUNDARY IS A CONTINUUM)
 002607 914 FORMAT (//,39H THE RIGHT BOUNDARY IS A FREE SURFACE)
 002607 915 FORMAT (//,60H THE RIGHT BOUNDARY IS A PISTON WITH INITIAL VELOC
 ITY OF ,1PE10.3,24H AND FINAL VELOCITY OF ,1PE10.3,/,10X,23H WHEN
 2 W+3 IS LESS THAN ,1PE10.3, 6H WITH ,15,44H CELLS INITIALLY AND A
 3 DDING ON A CELL EVERY ,15, 7H CYCLES)
 002607 916 FORMAT (//,50H THE RIGHT BOUNDARY IS A PISTON WITH VELOCITY = ,
 11PE10.3,1H+.1PE10.3,6H(TIME),/,10X. 5H WITH ,15,44H CELLS INITIALLY
 2 AND ADDING ON A CELL EVERY ,15, 7H CYCLES)
 002607 917 FORMAT (////,30X,33H THE PARAMETERS FOR EACH COMPONENT)
 002607 918 FORMAT (////,21H * THE COMPONENT IS ,2A6,/,11H AND HAS ,15,
 122H SPACE INCREMENTS OF ,1PE10.3,34H CM EACH AND A TIME INCREMENT
 2 OF ,1PE10.3,13H MICROSECONDS)
 002607 919 FORMAT (//,80H IT IS TO BE CONSIDERED AS AN EXPLOSIVE AND WILL B
 E BURNED BY THE TECHNIQUE OF)
 002607 921 FORMAT (//,53H THE ARRHENIUS RATE LAW WITH ACTIVATION ENERGY OF
 1 ,1PE10.3,46H CALORIES PER MOLE AND A FREQUENCY FACTOR OF ,/,3X,
 21PE10.3,16H PER MICROSECOND)
 002607 922 FORMAT (//,45H THE CJ VOLUME BURN WITH A BURN VOLUME OF ,1PE10.
 13,17H AND A MAX W OF ,1PE10.3)
 002607 923 FORMAT (//,68H THE ARTIFICIAL PIC TYPE OF VISCOSITY IS USED WITH
 1 A CONSTANT OF ,1PE10.3)
 002607 924 FORMAT (//,74H THE ARTIFICIAL LANDSHOFF TYPE OF VISCOSITY IS USE
 D WITH A CONSTANT OF ,1PE10.3)
 002607 925 FORMAT (//,76H THE TRUE FORM OF VISCOSITY IS USED WITH THE COEFF
 ICIENT OF VISCOSITY OF ,1PE10.3)
 002607 926 FORMAT (//,43H THE INITIAL CONDITIONS ARE A DENSITY OF ,1PE10.3,
 123H GRAMS/CC. PRESSURE OF ,1PE10.3,26H MEGABARS, TEMPERATURE OF ,/
 2,2X,1PE10.3,37H DEGREES KELVIN, INTERNAL ENERGY OF ,1PE10.3,33H M
 3B-CC/GM, PARTICLE VELOCITY OF ,1PE10.3,16H CM/MICROSECOND,/,37H
 4 AND MASS FRACTION OF THE SOLID OF ,1PE10.3)
 002607 927 FORMAT (//129H THE HOM EQUATION OF STATE PARAMETERS FOR THE SOLI
 D C,S,VS,C1,S1,FS,GS,HS,IS,JS,GG,CV,V0,ALPHA,SPA,USP,T0,P0,Y0,MU,P
 2LAP,SPP,VS2,/,6(2X,1PE18,11))
 002607 928 FORMAT (//108H THE HOM EQUATION OF STATE PARAMETERS FOR THE DETO
 NATION PRODUCTS A,B,C,D,E, K,L,M,N,O, Q,R,S,T,U,CVG,Z,/,6(2X,
 21PE18,11))
 002607 929 FORMAT (//,57H A GAMMA LAW TAYLOR WAVE WITH DETONATION VELOCITY
 10F ,1PE10.3)
 002607 930 FORMAT (//,77H THE LEFT BOUNDARY IS A STEADY STATE PISTON WITH A
 1 DETONATION VELOCITY OF ,1PE10.3,17H SPIKE VOLUME OF ,1PE10.3,/,
 26H WITH ,15,44H CELLS INITIALLY AND ADDING ON A CELL EVERY ,15,
 3 7H CYCLES)
 002607 END

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SUBROUTINE BURN (NMIN,NMAX)
C
C   BURN PERFORMS THE CHEMICAL REACTION ON CELLS WITH INDICES
C   NMIN TO NMAX
C
000004   COMMON CR(999),CRO(999),CU(999),CUO(999),CV(999),CVO(999),
1CM(999),CQ(999),ICF(999),CW(999),CWO(999),CT(999),CP(999),
2CS(999),CSZ(999),CI(999),CIO(999),LABEL(12)
000004   COMMON /MISC/ TTIME,DELT,ICYCL,PCNT,GCNT,DCNT,DINC,PINC,GINC,
1IALPH,HEATC,GASW,NCL,PTMIN,IPNT,IMC,IPR,ITR,IWR,IIR,IUR,IHC,ILB,
2IRB,WSW,PSMIN,WMAX,SMIN,ALB,BLB,ARB,BRB,NLI,NRI,NLINC,NRINC,
3NLCNT,NRCNT,NLH,NRL,DX,KIND,RXZDV,RXZVS,FPO,FEO,RXZ
000004   COMMON /MATER/ E(10),GAS(17,10),IBRN(10),SOL(23,10),VCJ(10),
1VFACT(10),Z(10),NAM(2,10),DTIME(10),NOINC(10),IVIS(10),IEXP(10),
2NCOM
000004   DATA WMAX /.99/
C   SET W TO 0 IF LESS THAN GASW
000004   DATA GASW /*0.02/
000004   DO 10 I=NMIN,NMAX
C   SKIP IF ALL BURNED
000005   IF (CW(I).EQ.0.) GO TO 10
C   J IS INDEX OF COMPONENT
000007   KSP=ICF(I)/64
000012   J=ICF(I)-64*KSP
C   CHECK FOR EXPLOSIVE OR GAS
000016   IF (IEXP(J).EQ.0) GO TO 10
C   DETERMINE TYPE OF BURN
000020   IF (IBRN(J).NE.0) GO TO 1
C   ARRHENIUS BURN
C   IF E,Z, OR T LESS THAN 0.0001 DO NOT BURN
000021   IF (E(J).LT.0.0001) GO TO 2
000024   IF (Z(J).LT.0.0001) GO TO 2
000026   IF (CT(I).LT.0.0001) GO TO 2
000031   CW(I)=CWO(I)*(1.-DELT*Z(J)*EXP(-E(J)/(1.9865*CT(I))))
000050   GO TO 2
C   CJ VOLUME BURN
000050   1 CW(I)=1.-(SOL(13,J)-CV(I))/(SOL(13,J)-VCJ(J))
000064   IF (CW(I).GT.CWO(I)) CW(I)=CWO(I)
000072   2 IF (CW(I).LT.GASW) CW(I)=0.
000077   10 CONTINUE
000102   RETURN
000102   END

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SUBROUTINE ENERGY (NMIN,NMAX)
C
C ENERGY COMPUTES THE NEW INTERNAL ENERGY FOR CELLS WITH INDICES
C NMIN TO NMAX
C
000004 COMMON CR(999),CRO(999),CU(999),CUO(999),CV(999),CVO(999),
1CM(999),CQ(999),ICF(999),CW(999),CWO(999),CT(999),CP(999),
2CS(999),CSZ(999),CI(999),CIO(999),LABEL(12)
000004 COMMON /MISC/ TTIME,DELTA,ICYCL,PCNT,GCNT,DCNT,DINC,PINC,GINC,
1IALPH,HEATC,GASW,NCL,PTMIN,IPNT,IMC,IPT,IWR,IIR,IUR,IHC,ILB,
2IRB,WSW,PSMIN,WMAX,SMIN,ALB,BLB,ARB,BRB,NLI,NRI,NLINC,NRINC,
3NLCNT,NRCNT,NLH,NRL,DX,KIND,RXZDV,RXZVS,FPO,FEO,RXZ
000004 COMMON /MATER/ E(10),GAS(17,10),IBRN(10),SOL(23,10),VCJ(10),
1VFAC(10),Z(10),NAM(2,10),DTIME(10),NOINC(10),IVIS(10),IEXP(10),
2NCOM
000004 IF (IALPH.EQ.0) GO TO 15
000005 DO 10 I=NMIN,NMAX
000006 CI(I)=CIO(I)+DELTA/CM(I)*(((CM(I)*CP(I-1)+CM(I-1)*CP(I))/
1(CM(I)+CM(I-1))+.5*(CQ(I)+CQ(I-1)))*CU(I)*CR(I)**IALPH-((CM(I+1)*
2CP(I)+CM(I)*CP(I+1))/(CM(I)+CM(I+1))+.5*(CQ(I)+CQ(I+1)))*CU(I+1)
3*CR(I+1)**IALPH)+((CUO(I+1)+CUO(I))**2-(CU(I+1)+CU(I))**2)*.125
000077 10 CONTINUE
000102 RETURN
000102 15 DO 20 I=NMIN,NMAX
000104 CI(I)=CIO(I)+DELTA/CM(I)*(((CM(I)*CP(I-1)+CM(I-1)*CP(I))/
1(CM(I)+CM(I-1))+.5*(CQ(I)+CQ(I-1)))*CU(I)-((CM(I+1)*
2CP(I)+CM(I)*CP(I+1))/(CM(I)+CM(I+1))+.5*(CQ(I)+CQ(I+1)))*CU(I+1)
3+((CUO(I+1)+CUO(I))**2-(CU(I+1)+CU(I))**2)*.125
000162 20 CONTINUE
000164 RETURN
000164 END

```

```

SUBROUTINE EQST (NMIN,NMAX)
C
C EQST DOES THE ELASTIC PLASTIC SPALLING EQUATION OF STATE
C CALCULATION FOR CELLS WITH INDICES NMIN TO NMAX
C
000004 COMMON CR(999),CRO(999),CU(999),CUO(999),CV(999),CVO(999),
1CM(999),CQ(999),ICF(999),CW(999),CWO(999),CT(999),CP(999),
2CS(999),CSZ(999),CI(999),CIO(999),LABEL(12)
000004 COMMON /MISC/ TTIME,DELT,ICYCL,PCNT,GCNT,DCNT,DINC,PINC,GINC,
1IALPH,HEATC,GASW,NCL,PTMIN,IPNT,IMC,IPR,ITR,IWR,IIR,IUR,IHC,ILB,
2IRB,WSW,PSMIN,WMAX,SMIN,ALB,BLB,ARB,BRB,NLI,NRI,NLINC,NRINC,
3NLCNT,NRCNT,NLH,NRL,DX,KIND,RXZDV,RXZVS,FPO,FEO,RXZ
000004 COMMON /MATER/ E(10),GAS(17,10),IBRN(10),SOL(23,10),VCJ(10),
1VFACT(10),Z(10),NAM(2,10),DTIME(10),NOINC(10),IVIS(10),IEXP(10),
2NCOM
000004 DIMENSION EQ(5)
C PSMIN = MINIMUM ELASTIC-PLASTIC PRESSURE
000004 DATA PSMIN /1.E-4/
000004 RFRNT=0.
000005 DO 20 I=NMIN,NMAX
C KSP = 0 FOR NON SPALLED CELL
C = 1 FOR SPALLED CELL
C = 2 FOR INTERFACE CELL
000006 KSP=ICF(I)/64
C J IS INDEX OF COMPONENT
000012 J=ICF(I)-64*KSP
000015 IF (KSP.EQ.2) RFRNT=CR(I)
000021 IF (CV(I).LT.SOL(13,J)) GO TO 1
000027 IF (KSP.NE.0) GO TO 11
000030 1 EQ(1)=CV(I)
000033 EQ(2)=CI(I)
C CHECK FOR CJ EQUATION OF STATE
000035 IF (IBRN(J).NE.0) GO TO 9
000037 EQ(3)=CW(I)
000041 EQ(4)=0.
000042 IF (CV(I).GT.SOL(13,J)) EQ(4)=(CP(I))/ABS(CR(I)-RFRNT)
000056 CALL HOM (EQ,SOL(1,J),GAS(1,J),INDH)
000072 IF (INDH) 2,4,3
C ERROR IN HOM
000073 2 PRINT 900,I,EQ
C MAKE FAKE CALL TO TIME TO GET A LAST PRINT
000105 PCNT=1.
000106 GCNT=100.
000107 DCNT=100.
000110 ICYCL=ICYCL-1
000111 TTIME=TTIME-DELT
000113 CALL TIME
000117 STOP
000121 900 FORMAT (16H1 HOM ERROR ,I5,5E18.11)
C CELL SPALLED
000121 3 IF (ICF(I).LT.64) ICF(I)=ICF(I)+64
000127 RFRNT=CR(I)
000132 CP(I)=EQ(4)
000134 GO TO 11
C REGULAR HOM EXIT
000134 4 CP(I)=EQ(4)
000137 IF (ABS(EQ(4)).LT.SOL(18,J)) CP(I)=SOL(18,J)
000151 CT(I)=EQ(5)
C IF AN EXPLOSIVE OR GAS, SKIP ELASTIC-PLASTIC
000154 IF (IEXP(J).NE.0) GO TO 20
C IF YO OR MU LESS THAN 0.0001 SKIP ELASTIC CALCULATION

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```

SUBROUTINE HETCON (NMIN,NMAX)
C
C   HETCON COMPUTES THE ADDITION TO THE INTERNAL ENERGY DUE TO
C   HEAT CONDUCTION FOR CELLS WITH INDICES NMIN TO NMAX
C
000004   COMMON CR(999),CRO(999),CU(999),CUO(999),CV(999),CVO(999),
1CM(999),CQ(999),ICF(999),CW(999),CWO(999),CT(999),CP(999),
2CS(999),CSZ(999),CI(999),CIO(999),LABEL(12)
000004   COMMON /MISC/ TTIME,DELT,ICYCL,PCNT,GCNT,DCNT,DINC,PINC,GINC,
1IALPH,HEATC,GASW,NCL,PTMIN,IPNT,IMC,IPR,ITR,IWR,IIR,IUR,IHC,ILB,
2IRB,WSW,PSMIN,WMAX,SMIN,ALB,BLB,ARB,BRB,NLI,NRI,NLINC,NRINC,
3NLCNT,NRCNT,NLH,NRL,DX,KIND,RXZDV,RXZVS,FPO,FEO,RXZ
000004   COMMON /MATER/ E(10),GAS(17,10),IBRN(10),SOL(23,10),VCJ(10),
1VFACT(10),Z(10),NAM(2,10),DIME(10),NOINC(10),IVIS(10),IEXP(10),
2NCOM
000004   IF (IALPH.EQ.0) GO TO 15
000005   DO 10 I=NMIN,NMAX
000006   10 CI(I)=CI(I)+2.*HEATC*DELT/CM(I)*((CR(I+1)**IALPH)*(CT(I+1)-CT(I))
1/(CR(I+2)-CR(I))-(CR(I)**IALPH)*(CT(I)-CT(I-1))/(CR(I+1)-CR(I-1)))
000047   RETURN
000047   15 DO 20 I=NMIN,NMAX
000051   20 CI(I)=CI(I)+2.*HEATC*DELT/CM(I)*((CT(I+1)-CT(I))
1/(CR(I+2)-CR(I))-(CT(I)-CT(I-1))/(CR(I+1)-CR(I-1)))
000077   RETURN
000077   END

```

```

SUBROUTINE RADIUS (NMIN,NMAX)
C
C   RADIUS COMPUTES THE CHANGES IN THE R COORDINATE IN A TIME STEP
C   FOR CELLS WITH INDICES NMIN TO NMAX
C
000004   COMMON CR(999),CRO(999),CU(999),CUO(999),CV(999),CVO(999),
1CM(999),CQ(999),ICF(999),CW(999),CWO(999),CT(999),CP(999),
2CS(999),CSZ(999),CI(999),CIO(999),LABEL(12)
000004   COMMON /MISC/ TTIME,DELT,ICYCL,PCNT,GCNT,DCNT,DINC,PINC,GINC,
1IALPH,HEATC,GASW,NCL,PTMIN,IPNT,IMC,IPR,ITR,IWR,IIR,IUR,IHC,ILB,
2IRB,WSW,PSMIN,WMAX,SMIN,ALB,BLB,ARB,BRB,NLI,NRI,NLINC,NRINC,
3NLCNT,NRCNT,NLH,NRL,DX,KIND,RXZDV,RXZVS,FPO,FEO,RXZ
000004   COMMON /MATER/ E(10),GAS(17,10),IBRN(10),SOL(23,10),VCJ(10),
1VFACT(10),Z(10),NAM(2,10),DTIME(10),NOINC(10),IVIS(10),IEXP(10),
2NCOM
000004   DO 10 I=NMIN,NMAX
000005   10 CR(I)=CRO(I)+CU(I)*DELT
000014   RETURN
000014   END

```



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SUBROUTINE STRESS (NMIN,NMAX)
C
C STRESS CALCULATES THE CHANGE IN THE STRESS DEVIATOR FOR CELLS
C WITH INDICES NMIN TO NMAX
C
000004 COMMON CR(999),CRO(999),CU(999),CUO(999),CV(999),CVO(999),
1CM(999),CQ(999),ICF(999),CW(999),CWO(999),CT(999),CP(999),
2CS(999),CSZ(999),CI(999),CIO(999),LABEL(12)
000004 COMMON /MISC/ TTIME,DELT,ICYCL,PCNT,GCNT,DCNT,DINC,PINC,GINC,
1IALPH,HEATC,GASW,NCL,PTMIN,IPNT,IMC,IPR,ITR,IWR,IIR,IUR,IHC,ILB,
2IRB,WSW,PSMIN,WMAX,SMIN,ALB,BLB,ARB,BRB,NLI,NRI,NLINC,NRINC,
3NLCNT,NRCNT,NLH,NRL,DX,KIND
000004 COMMON /MATER/ E(10),GAS(17,10),IBRN(10),SOL(23,10),VCJ(10),
1VFACT(10),Z(10),NAM(2,10),DTIME(10),NOINC(10),IVIS(10),IEXP(10),
2ZNCOM
C CHANGE IN S MUST BE GREATER THAN SMIN BEFORE AFFECTING ANYTHING
000004 DATA SMIN /1.E-9/
000004 DO 10 I=NMIN,NMAX
000005 KSP=ICF(I)/64
C J IS INDEX OF COMPONENT
000011 J=ICF(I)-64*KSP
C IF YO OR MU LESS THAN 0.0001 SKIP ELASTIC CALCULATION
000014 IF(SOL(19,J).LT.0.0001) GO TO 10
000021 IF(SOL(20,J).LT.0.0001) GO TO 10
000024 DELS=2.*SOL(20,J)*((CU(I)-CU(I+1))*DELT/(CR(I+1)-CR(I))
1+.6666666667*(CV(I)-CVO(I))/(CV(I)+CVO(I)))
000047 IF (ABS(DELS).LT.SMIN) DELS=0.
000053 CS(I)=CS(I) + DELS
000057 IF (IALPH.EQ.1) GO TO 20
000061 GO TO 10
C CALCULATE CSZ
000061 20 DELS = SOL(20,J)*1.33333334*(CV(I)-CVO(I))/(CV(I)+CVO(I))
000074 IF (ABS(DELS).LT.SMIN) DELS = 0.
000100 CSZ(I) = CSZ(I) + DELS
000104 10 CONTINUE
000107 RETURN
000107 END

```

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SUBROUTINE TIME
C
C TIME INCREMENTS THE TIME FOR EACH CYCLE AND DOES THE
C PRINTING AND GRAPHING WHEN NECESSARY
C
000002 COMMON CR(999),CRO(999),CU(999),CUO(999),CV(999),CVO(999),
1CM(999),CQ(999),ICF(999),CW(999),CWO(999),CT(999),CP(999),
2CS(999),CSZ(999),CI(999),CIO(999),LABEL(12)
000002 COMMON /MISC/ TTIME,DELT,ICYCL,PCNT,GCNT,DCNT,DINC,PINC,GINC,
1IALPH,HEATC,GASW,NCL,PTMIN,IPNT,IMC,IPR,ITR,IWR,IIR,IUR,IHC,ILB,
2IRB,WSW,PSMIN,WMAX,SMIN,ALB,BLB,ARB,BRB,NLI,NRI,NLINC,NRINC,
3NLCNT,NRCNT,NLH,NRL,DX,KIND,RXZDV,RXZVS,FPO,FEO,RXZ
000002 COMMON /MATER/ E(10),GAS(17,10),IBRN(10),SOL(23,10),VCJ(10),
1VFACT(10),Z(10),NAM(2,10),DTIME(10),NOINC(10),IVIS(10),IEXP(10),
2NCOM
000002 DIMENSION ICHAR(2),TKE(10),TIE(10)
C PLOT CHARACTERS 1. . 2. X
000002 DATA ICHAR(1) /052/
000002 DATA ICHAR(2) /067/
C THE PRESSURE OF ONE CELL OF A COMPONENT MUST BE GREATER THAN
C PTMIN TO USE THAT COMPONENTS DELTA T
000002 DATA PTMIN /1.E-05/
000002 DATA ICYCL /0/
000002 DATA PCNT /1./
000002 DATA GCNT /1./
000002 DATA DCNT /100./
C INCREMENT TIME AND CYCLE NUMBER
000002 ICYCL=ICYCL+1
000004 TTIME=TTIME+DELT
000006 NRCNT=NRCNT-1
000007 NLCNT=NLCNT-1
C SEE IF WE NEED TO ADD ON CELLS FROM LEFT
000010 IF (NLCNT.GT.0) GO TO 4
C YES ADD ON A CELL AND SET COUNT
000012 NLH=NLH+1
000013 NLCNT=NLINC
C SEE IF WE NEED TO ADD ON CELLS FROM RIGHT
000014 4 IF (NRCNT.GT.0) GO TO 5
C YES ADD ON A CELL AND SET COUNT
000017 NRL=NRL-1
000020 NRCNT=NRINC
C CHECK IF ALL CELLS ARE IN THE CALCULATION
000021 5 IF (NRL.GT.NLH+1) GO TO 6
C YES SET NLI,NRI,NLH, AND COUNTS ACCORDINGLY
000026 NLH=NCL+1
000027 NLI=1
000030 NRI=0
000031 NLCNT=100000
000032 NRCNT=100000
000033 GO TO 7
C CHECK IF ALL CELLS ARE IN THE CALCULATION
000033 6 IF (NLH.LT.NCL) GO TO 7
C YES SET NLI,NRI,NLH, AND COUNTS ACCORDINGLY
000035 NRCNT=100000
000037 NLCNT=100000
000040 NLH=NCL+1
000041 NRI=0
000042 NLI=1
C DETERMINE DELT FOR NEXT CYCLE
000043 7 K=1
C SAVE DELTA T FOR PRINT OUT

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000044      SVDEL=DELT
000046      DO 10 I=1,NCOM
000047      N=NOINC(I)
000051      DO 8 J=1,N
000053      IF (CP(K+J).GT.PTMIN) GO TO 9
000060      8 CONTINUE
000062      GO TO 10
000062      9 IF (DELT.GT.DTIME(I)) DELT=DTIME(I)
000067      10 K=K+N
C
000073      DECREMENT COUNTS
000075      PCNT=PCNT-1.
000075      GCNT=GCNT-1.
000077      DCNT=DCNT-1.
000100      IF (PCNT.GT.0.) GO TO 20
000102      PCNT=PINC
000103      PRINT 900,ICYCL,TTIME,LABEL
000114      PRINT 906,SVDEL,NLH,NRL
000126      LIN=56
000127      K=1
000130      DO 15 I=1,NCOM
000132      N=NOINC(I)
000134      TKE(I)=0.
000136      TIE(I)=0.
000137      L=K
000141      DO 11 J=1,N
000142      TKE(I)=TKE(I)+CM(L)*CU(L+1)*CU(L+1)
000150      TIE(I)=TIE(I)+CM(L)*CI(L)
000154      L=L+1
000155      11 CONTINUE
000160      TKE(I)=TKE(I)*.50036186
000163      IF (LIN.LT.6) GO TO 12
000165      PRINT 901 ,NAM(1,I),NAM(2,I),TKE(I),TIE(I)
000206      LIN=LIN-5
000210      GO TO 13
000210      12 PRINT 902,NAM(1,I),NAM(2,I),TKE(I),TIE(I)
000232      PRINT 903
000236      LIN=55
000237      13 IF (I.EQ.1) PRINT 903
000245      IF (I.EQ.1) N=N+1
000250      IF (I.EQ.NCOM) N=N+1
000253      EINO=-1.
000255      DO 14 J=1,N
000256      IF (CI(K).EQ.EINO) GO TO 14
000261      IFLG=ICF(K)/64
000264      IF (LIN.EQ.0) PRINT 904
000271      IF (LIN.EQ.0) LIN=58
000273      PRINT 905,K,CR(K),CU(K),CV(K),CI(K),CP(K),CT(K),CW(K),CQ(K),CM(K),
1CS(K),CSZ(K),IFLG
000344      LIN=LIN-1
000346      EINO=CI(K)
000350      14 K=K+1
000354      15 CONTINUE
000357      IF (IMC.EQ.0) GO TO 20
C
C
C      4020 OUTPUT BELOW
C
000360      WRITE (12,900) ICYCL,TTIME,LABEL
000371      WRITE (12,906) SVDEL,NLH,NRL
000403      LIN=56
000404      K=1
000405      DO 19 I=1,NCOM
000407      EINO=-1.

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000411      N=NOINC(I)
000413      IF (LIN.LT.6) GO TO 16
000415      WRITE (12,901) NAM(1,I),NAM(2,I).TKE(I).TIE(I)
000436      LIN=LIN-5
000440      GO TO 17
000440      16 WRITE (12,902) NAM(1,I),NAM(2,I).TKE(I).TIE(I)
000462      WRITE (12,903)
000466      LIN=55
000467      17 IF (I.EQ.1) WRITE (12,903)
000475      IF (I.EQ.1) N=N+1
000500      IF (I.EQ.NCOM) N=N+1
000503      DO 18 J=1,N
000505      IF (CI(K).EQ.EINO) GO TO 18
000510      IFLG=ICF(K)/64
000513      IF (LIN.EQ.0) WRITE (12,904)
000520      IF (LIN.EQ.0) LIN=58
000522      WRITE (12,905) K,CR(K),CU(K),CV(K).CI(K),CP(K).CT(K),CW(K).CQ(K),
      LCM(K),CS(K),CSZ(K),IFLG
000573      LIN=LIN-1
000575      EINO=CI(K)
000577      18 K=K+1
000603      19 CONTINUE
      C *****
      C          4020 OUTPUT ABOVE
      C *****
000606      20 IF (GCNT.GT.0.) GO TO 50
000611      GCNT=GINC
000612      IF (IMC.EQ.0) GO TO 50
      C *****
      C          4020 OUTPUT BELOW
      C *****
000613      IF (IPR.EQ.0) GO TO 32
      C GRAPH PRESSURE VS RADIUS
000614      CALL ADV(1)
000615      CALL DGA (123,1023,0,900.0.,CR(NCL+1),.90,-.1)
000625      CALL DLNLN (10,10)
000627      CALL SLLIN (10,2)
000631      CALL SBLIN (10,3)
000633      J=1
000634      K=2
000635      DO 31 I=1,NCOM
000637      N=NOINC(I)
000641      CALL PLOT (N.CR(K).1.CP(K).1,ICHAR(J).1)
000652      J=3-J
000654      K=K+N
000655      31 CONTINUE
000660      CALL LINCNT (60)
000661      WRITE (12,910) LABEL
000667      WRITE (12,911) TTIME,ICYCL
000677      32 IF (ITR.EQ.0) GO TO 34
      C GRAPH TEMPERATURE VS RADIUS
000700      CALL ADV(1)
000702      CALL DGA (123,1023,0,900.0.,CR(NCL+1),5000.,0.)
000712      CALL DLNLN (10,10)
000714      CALL SLLIN (10,0)
000716      CALL SBLIN (10,3)
000720      J=1
000721      K=2
000722      DO 33 I=1,NCOM
000724      N=NOINC(I)
000726      CALL PLDT (N.CR(K).1.CT(K).1,ICHAR(J).1)
000737      J=3-J

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000741      K=K+N
000742      33 CONTINUE
000743      CALL LINCNT (60)
000744      WRITE (12,910) LABEL
000745      WRITE (12,912) TTIME,ICYCL
000754      34 IF (IWR.EQ.0) GO TO 36
000764      C      GRAPH MASS FRACTION VS RADIUS
000765      CALL ADV(1)
000767      CALL DGA (123,1023,0,900,0.,CR(NCL+1),1.,0.)
000777      CALL DLNLN (10,10)
001001      CALL SLLIN (10,2)
001003      CALL SBLIN (10,3)
001005      J=1
001006      K=2
001007      DO 35 I=1,NCOM
001011      N=NOINC(I)
001013      CALL PLOT (N,CR(K),1,CW(K),1,ICHAR(J),1)
001024      J=3-J
001026      K=K+N
001027      35 CONTINUE
001032      CALL LINCNT (60)
001033      WRITE (12,910) LABEL
001041      WRITE (12,913) TTIME,ICYCL
001051      36 IF (IIR.EQ.0) GO TO 38
001052      C      GRAPH VOLUME VS RADIUS
001054      CALL ADV(1)
001064      CALL DGA (123,1023,0,900,0.,CR(NCL+1),2.,0.)
001066      CALL DLNLN (10,10)
001070      CALL SLLIN (10,3)
001072      CALL SBLIN (10,3)
001073      J=1
001074      K=2
001076      DO 37 I=1,NCOM
001077      N=NOINC(I)
001100      CALL PLOT (N,CR(K),1,CV(K),1,ICHAR(J),1)
001111      J=3-J
001113      K=K+N
001114      37 CONTINUE
001117      CALL LINCNT (60)
001120      WRITE (12,910) LABEL
001126      WRITE (12,914) TTIME,ICYCL
001136      38 IF (IUR.EQ.0) GO TO 50
001137      C      GRAPH VELOCITY VS RADIUS
001141      CALL ADV(1)
001151      CALL DGA (123,1023,0,900,0.,CR(NCL+1),1.0,-1.0)
001153      CALL DLNLN (10,10)
001155      CALL SLLIN (10,2)
001157      CALL SBLIN (10,3)
001160      J=1
001161      K=2
001163      DO 39 I=1,NCOM
001165      N=NOINC(I)
001176      CALL PLOT (N,CR(K),1,CU(K),1,ICHAR(J),1)
001200      J=3-J
001201      K=K+N
001204      39 CONTINUE
001205      CALL LINCNT (60)
001213      WRITE (12,910) LABEL
001213      WRITE (12,915) TTIME,ICYCL
001213      C      *****
001213      C      4020 OUTPUT ABOVE
001213      C      *****

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001223      50 IF (DCNT.GT.0.) RETURN
001225      DCNT=DINC
001227      WRITE (1)CRO,CUO,CWO,CSZ,CIO,CVO,ICF,CM,CQ,CT,CP,LABEL,DELT,HEATC,
1IALPH,TTIME,PINC,GINC,DINC,PCNT,GCNT,DCNT,E,GAS,IBRN,SOL,VCJ,
2VFACT,Z,NAM,DTIME,NOINC,NCOM,ICYCL,NCL,IPNT,IMC,IPR,ITR,IWR,IIR,
3IUR,IHC,ILB,IRB,ALB,BLB,ARB,BRB,WSW,NLI,NRI,NLINC,NRINC,NLCNT,
4NRCNT,NLH,NRL,DX,IVIS,IEXP,KIND,CR,CI,CW,CV,CU,CS,RXZDV,RXZVS,
5RXZ,FPO,FEO
001454      PRINT 916,ICYCL
001462      RETURN
001463      900 FORMAT (12H1 AT CYCLE ,I5.13H THE TIME IS ,1PE11.4,18H MICROSECON
1DS FOR ,12A6)
001463      901 FORMAT (//,20H THE COMPONENT IS ,2A6.26H WITH A KINETIC ENERGY 0
1F ,1PE11.4,27H AND AN INTERNAL ENERGY OF ,1PE11.4,/)
001463      902 FORMAT (20H1 THE COMPONENT IS ,2A6.26H WITH A KINETIC ENERGY 0
1F ,1PE11.4,27H AND AN INTERNAL ENERGY OF ,1PE11.4,/)
001463      903 FORMAT (129H CELL RADIUS VEL VOLUME ENERGY
1 PRESS TEMP MASS F VISC MASS X,Z ST
2RESS FLAG)
001463      904 FORMAT (129H1CELL RADIUS VEL VOLUME ENERGY
1 PRESS TEMP MASS F VISC MASS X,Z ST
2RESS FLAG)
001463      905 FORMAT(15,9(X,1PE11.4),X,2(3PF6.3).X,02)
001463      906 FORMAT (14H DELTA T IS ,1PE10.3,25H WITH LEFT CELL INDEX OF ,I5,
125H AND RIGHT CELL INDEX OF ,I5)
001463      910 FORMAT (10X,12A6)
001463      911 FORMAT (//,10X,38HPRESSURE (MEGABARS) VS RADIUS (CM) AT ,1PE10.3,
131H MICROSECONDS AND CYCLE NUMBER ,I5)
001463      912 FORMAT (//,10X,43HTEMPERATURE (DEG KELVIN) VS RADIUS (CM) AT ,
11PE10.3,31H MICROSECONDS AND CYCLE NUMBER ,I5)
001463      913 FORMAT (//,10X,41HMASS FRACTION OF SOLID VS RADIUS (CM) AT ,1PE10.
13,31H MICROSECONDS AND CYCLE NUMRER ,I5)
001463      914 FORMAT (//,10X,45H SPECIFIC VOLUME (CC/GM) VS RADIUS (CM) AT .
11PE10.3,31H MICROSECONDS AND CYCLE NUMBER ,I5)
001463      915 FORMAT (//,10X,46HPARTICLE VELOCITY (CM/MSEC) VS RADIUS (CM) AT ,
11PE10.3,31H MICROSECONDS AND CYCLE NUMBER ,I5)
001463      916 FORMAT (///,10X,19HTAPE DUMP AT CYCLE ,I5)
001463      END

```

```

SUBROUTINE UPDATE (NMIN,NMAX)
C
C
C
000004   COMMON CR(999),CRO(999),CU(999),CUO(999),CV(999),CVO(999),
1CM(999),CQ(999),ICF(999),CW(999),CWO(999),CT(999),CP(999),
000004   ZCS(999),CSZ(999),CI(999),CIO(999),LABEL(12)
COMMON /MISC/ TTIME,DELT,ICYCL,PCNT,GCNT,DCNT,DINC,PINC,GINC,
1IALPH,HEATC,GASW,NCL,PTMIN,IPNT,IMC,IPR,ITR,IWR,IIR,IUR,IHC,ILB,
2IRB,WSW,PSMIN,WMAX,SMIN,ALB,BLB,ARB,BRB,NLI,NRI,NLINC,NRINC,
3NLCNT,NRCNT,NLH,NRL,DX,KIND,RXZDV,RXZVS,FPO,FEO,RXZ
000004   COMMON /MATER/ E(10),GAS(17,10),IBRN(10),SOL(23,10),VCJ(10),
1VFACT(10),Z(10),NAM(2,10),DIME(10),NOINC(10),IVIS(10),IEXP(10),
2NCOM
000004   DO 10 I=NMIN,NMAX
000005       CRO(I)=CR(I)
000010       CUO(I)=CU(I)
000012       CVO(I)=CV(I)
000015       CWO(I)=CW(I)
000017   10  CIO(I)=CI(I)
000024       RETURN
000024       END

```

```

SUBROUTINE VELOC (NMIN,NMAX)
C
C VELOC CALCULATES THE NEW VELOCITY AT THE NEXT TIME STEP FOR
C CELLS WITH INDICES FROM NMIN TO NMAX
C
000004 COMMON CR(999),CRO(999),CU(999),CUO(999),CV(999),CVO(999),
1CM(999),CQ(999),ICF(999),CW(999),CWO(999),CT(999),CP(999),
2CS(999),CSZ(999),CI(999),CIO(999),LABEL(12)
000004 COMMON /MISC/ TTIME,DELT,ICYCL,PCNT,GCNT,DCNT,DINC,PINC,GINC,
1IALPH,HEATC,GASW,NCL,PTMIN,IPNT,IMC,IPR,ITR,IWR,IIR,IUR,IHC,ILB,
2IRB,WSW,PSMIN,WMAX,SMIN,ALB,BLB,ARB,BRB,NLI,NRI,NLINC,NRINC,
3NLCNT,NRCNT,NLH,NRL,DX,KIND,RXZDV,RXZVS,FPO,FEO,RXZ
000004 CDMON /MATER/ E(10),GAS(17,10),IBRN(10),SOL(23,10),VCJ(10),
1VFACT(10),Z(10),NAM(2,10),DIME(10),NOINC(10),IVIS(10),IEXP(10),
2NCOM
000004 IF (IALPH.EQ.0) GO TO 15
000005 DO 10 I=NMIN,NMAX
C
000006 J IS INDEX OF COMPONENT
000012 KSP=ICF(I)/64
000015 J = ICF(I)-64*KSP
CU(I)=CUO(I)+2.*DELT*(CRO(I)**IALPH)/(CM(I-1)+CM(I))*(CP(I-1)
1-CP(I)+CQ(I-1)-CQ(I))
C
000040 IF (SOL(19,J).LT.0.0001) GO TO 10
000045 IF (SOL(20,J).LT.0.0001) GO TO 10
000050 IF (CRO(I).EQ.0.) GO TO 10
000052 IF (IALPH.EQ.1) GO TO 50
C
000054 VELOCITY CALCULATION FOR ELASTIC SPHERE
000067 CU(I)=CU(I)-0.75*(CS(I)+CS(I-1))*(CVO(I)+CVO(I-1))*DELT/CRO(I)
GO TO 10
C
000067 VELOCITY CALCULATION FOR ELASTIC CYLINDER
50 CU(I)=CU(I)-(CS(I)+CS(I-1)+0.5*(CSZ(I)+CSZ(I-1)))*0.5*(CVO(I)+
1CVO(I-1))*DELT/CRO(I)
10 CONTINUE
000107 RETURN
000112
000112 15 DO 20 I=NMIN,NMAX
000114 20 CU(I)=CUO(I)+2.*DELT/(CM(I-1)+CM(I))*(CP(I-1)
1-CP(I)+CQ(I-1)-CQ(I))
000135 RETURN
000135 END

```



```

      SUBROUTINE VISCOS (NMIN,NMAX)
C
C   VISCOS COMPUTES THE VISCOSITY IN EITHER THE PIC, LANDSHOFF, OR
C   REAL FORM FOR CELLS WITH INDICES NMIN TO NMAX
C
000004   COMMON CR(999),CRO(999),CU(999),CUO(999),CV(999),CVO(999),
1CM(999),CQ(999),ICF(999),CW(999),CWO(999),CT(999),CP(999),
000004   2CS(999),CSZ(999),CI(999),CIO(999),LABEL(12)
      COMMON /MISC/ TTIME,DELT,ICYCL,PCNT,GCNT,DCNT,DINC,PINC,GINC,
1IALPH,HEATC,GASW,NCL,PTMIN,IPNT,IMC,IPR,ITR,IWR,IIR,IUR,IHC,ILB,
2IRB,WSW,PSMIN,WMAX,SMIN,ALB,BLB,ARB,BRB,NLI,NRI,NLINC,NRINC,
3NLCNT,NRCNT,NLH,NRL,DX,KIND,RXZDV,RXZVS,FPO,FEO,RXZ
000004   COMMON /MATER/ E(10),GAS(17,10),TBRN(10),SOL(23,10),VCJ(10),
1VFACT(10),Z(10),NAM(2,10),DTIME(10),NOINC(10),IVIS(10),IEXP(10),
2NCOM
000004   DO 10 I=NMIN,NMAX
000005   UT=CU(I)-CU(I+1)
000010   C   J IS INDEX OF COMPONENT
000013   KSP=ICF(I)/64
      J=ICF(I)-64*KSP
000017   C   CHECK FOR REAL FORM
000022   IF (IVIS(J).EQ.2) GO TO 2
000024   CQ(I)=0.
      IF (UT.LE.0.) GO TO 10
000025   C   CHECK FOR LANDSHOFF FORM
      IF (IVIS(J).EQ.1) GO TO 1
000030   C   PIC FORM
000042   CQ(I)=ABS(VFACT(J)*.5*(CU(I)+CU(I+1))*UT/CV(I))
      GO TO 10
000042   C   LANDSHOFF FORM
000050   1 CQ(I)=VFACT(J)*UT/CV(I)
      GO TO 10
000050   C   REAL FORM
000061   2 CQ(I)=1.3333*VFACT(J)*UT/(CV(I)*CM(I))
000064   10 CONTINUE
      RETURN
000064   END

```

```

SUBROUTINE VOLUM (NMIN,NMAX)
C
C
C
000004 COMMON CR(999),CRO(999),CU(999),CUO(999),CV(999),CVO(999),
1CM(999),CQ(999),ICF(999),CW(999),CWO(999),CT(999),CP(999),
2CS(999),CSZ(999),CI(999),CIO(999),LABEL(12)
000004 COMMON /MISC/ TTIME,DELT,ICYCL,PCNT,GCNT,DCNT,DINC,PINC,GINC,
1IALPH,HEATC,GASW,NCL,PTMIN,IPNT,IMC,IPR,ITR,IWR,IIR,IUR,IHC,ILB,
2IRB,WSW,PSMIN,WMAX,SMIN,ALB,BLB,ARB,BRB,NLI,NRI,NLINC,NRINC,
3NLCNT,NRCNT,NLH,NRL,DX,KIND,RXZDV,RXZVS,FPO,FEO,RXZ
000004 COMMON /MATER/ E(10),GAS(17,10),IBRN(10),SOL(23,10),VCJ(10),
1VFACT(10),Z(10),NAM(2,10),DIME(10),NOINC(10),IVIS(10),IEXP(10),
2NCOM
000004 DO 10 I=NMIN,NMAX
000005 10 CV(I)=((CR(I+1)+CR(I))*5)**IALPH*(CR(I+1)-CR(I))/CM(I)
000024 RETURN
000024 END

```



```

000006      DATA GASW /.01/
000006      DATA SOLW /.999/
000006      DATA SPMIN/5.0E-3/
000006      DATA VGSS /.9/
000006      DATA VIT(3) /1.E-5/
000006      DATA VIT(10) /0./
000006      IND=0
000006      IF (V(3).GT.SOLW) GO TO 10
000012      IF (V(3).LT.GASW) GO TO 110
000014      GO TO 210
C EQUATION OF STATE FOR SOLID ONLY
000014      10 IF (V(1).GT.S(13)) GO TO 50
C FOR TWO PHASE FE TYPE EQUATION OF STATE
000020      IF (V(1).GT.S(3)) GO TO 11
000023      IF (V(1).LT.S(23)) GO TO 45
000024      V(1)=S(23)
000026      GO TO 45
000026      11 C1=S(1)
000027      S1=S(2)
000031      20 VOMV=S(13)-V(1)
000033      HP=((C1/(S(13)-S1*VOMV))*2)*VOMV
000040      HE=HP*VOMV*.5
000042      V(4)=HP*(V(2)-HE)*S(11)/V(1)
C IF NO HEAT CAPACITY SKIP TEMP CALCULATION
000047      IF (S(12)) 21,22,21
000050      21 ALNV=ALOG(V(1))
000055      V(5)=(V(2)-HE)*23890./S(12)+EXP(S(6)+ALNV*(S(7)+ALNV*(S(8)+ALNV*
1(S(9)+ALNV*S(10))))
000077      22 RETURN
C SWITCH TO SECOND US. UP FIT
000100      45 C1=S(4)
000102      S1=S(5)
000103      GO TO 20
C SPALLING SOLID EQUATION OF STATE
000104      50 DPDX=V(4)
C IF ALPHA IS ZERO SET P=P0 AND DO NOT SPALL
000106      IF (S(14)) 51,51,52
000107      51 V(4)=S(18)
000111      V(5)=S(17)
000112      RETURN
000113      52 V(4)=(S(11)*(V(2)+(1.-V(1)/S(13))*S(12)*1.39528394E-5/S(14)))/V(1)
000125      V(5)=V(2)*23890./S(12)+S(17)
000131      IF (DPDX.GE.0.) RETURN
C IF SPA LESS THAN / 0.0001 DO NOT SPALL
000134      IF (S(15) .LT.0.0001) RETURN
000137      T=S(15)*SQRT(-DPDX)
000147      SPLP=-T
000150      IF (T.GT.S(16)) SPLP=-S(16)
000154      IF (T.LT.SPMIN) SPLP=-SPMIN
000160      IF (V(4).GT.SPLP) RETURN
000163      V(4)=S(22)
C SET IND FOR SPALLED SOLID
000165      IND=1
000166      RETURN
C EQUATION OF STATE FOR GAS ONLY
000166      110 ALNV=ALOG(V(1))
000173      ALNPI=G(1)+ALNV*(G(2)+ALNV*(G(3)+ALNV*(G(4)+ALNV*(G(5))))
000203      ALNII=G(6)+ALNPI*(G(7)+ALNPI*(G(8)+ALNPI*(G(9)+ALNPI*(G(10))))
000214      ALNTI=G(11)+ALNV*(G(12)+ALNV*(G(13)+ALNV*(G(14)+ALNV*(G(15))))
000225      EI=EXP(ALNII)-G(17)
000233      V(4)=EXP(ALNPI)*(EI-V(2))/V(1)*(G(12)+ALNV*(G(13)+G(13)+ALNV*(
13.*G(14)+ALNV*4.*G(15))))

```

```

000257      V(5)=EXP(ALNTI)*(V(2)-EI)*23890./G(16)
000272      RETURN
C EQUATION OF STATE FOR MIXTURE OF SOLID AND GAS
000272      210 OMW=1.-V(3)
000274      OMWR=1./OMW
000276      IF (V(1).LT.S(13)) GO TO 230
000300      WR=1./V(3)
000302      VIT(1)=(V(1)-V(3)*S(13)*VGSS)*OMWR
000307      VIT(2)=1.002
C      IBR=1 FOR ITERATION ON VG
000310      IBR=1
000311      215 CALL LFB (X,F,VIT)
000321      IF (VIT(10)) 900,260,220
000323      220 IF (X.LE.0.) GO TO 225
000325      VG=X
000326      VS=(V(1)-OMW*VG)*WR
000332      IF (VS.LE.0.) GO TO 225
000333      IF (VS.GT.S(13)) VS=S(13)
000337      GO TO 250
C      SET VS=VG=VOLUME WHEN GET IN TROUBLE
000340      225 VS=V(1)
000341      VG=V(1)
000342      X=V(1)
000343      GO TO 250
000344      230 VIT(1)=V(1)
000345      VIT(2)=.999
C      IBR=2 FOR ITERATION ON VS
000347      IBR=2
000350      235 CALL LFB (X,F,VIT)
000360      IF (VIT(10)) 900,260,240
000362      240 IF (X.LE.0.) GO TO 225
000364      IF (X.GT.S(13)) X=S(13)
000367      VS=X
000370      VG=(V(1)-V(3)*VS)*OMWR
000374      IF (VG.LE.0.) GO TO 225
C      CALCULATE TEMPERATURE/PRESSURE DIFFERENCE FOR MIXTURE ITERATION
000375      250 VOMV=S(13)-VS
000377      HP=((S(1)/(S(13)-S(2)*VOMV))*#2)*VOMV
000404      HE=HP*.5*VOMV
000406      ALNV=ALOG(VS)
000412      HT=EXP(S(6)+ALNV*(S(7)+ALNV*(S(8)+ALNV*(S(9)+ALNV*S(10))))
000427      ALNV=ALOG(VG)
000434      ALNPI=G(1)+ALNV*(G(2)+ALNV*(G(3)+ALNV*(G(4)+ALNV*G(5)))
000444      EI=EXP(G(6)+ALNPI*(G(7)+ALNPI*(G(8)+ALNPI*(G(9)+ALNPI*G(10))))-
1G(17)
000463      PI=EXP(ALNPI)
000467      TI=EXP(G(11)+ALNV*(G(12)+ALNV*(G(13)+ALNV*(G(14)+ALNV*G(15))))
000504      BETER=-G(12)+ALNV*(G(13)+G(13)+ALNV*(3.*G(14)+4.*ALNV*G(15)))
000515      TEMP=-G(16)*BETER/VG
000520      TEMP1=S(11)*S(12)/VS
000523      F=-(HT*TEMP1+TI*TEMP)*4.18585182E-5
000527      TEMP=TEMP+TEMP1
000531      VSTO=(S(12)-G(16))*V(3)+G(16)
000534      F=((OMW*G(16)*TI+V(3)*S(12)*HT)*4.18585182E-5+(EI-HE)*V(3)-E1+
1V(2))*TEMP/VSTO+F-PI+HP
000556      GO TO (215,235),IBR
C      HAVE FOUND A SOLUTION FOR THE MIXTURE
C      GET THE TEMPERATURE AND PRESSURE
000563      260 VARST=((((TI-HT)*G(16)*4.18585182E-5+V(2)*OMWR-EI)*S(12)+HE*G(16))
1*OMW/VSTO)-HE
000600      V(4)=HP+VARST*S(11)/VS
000604      V(5)=HT+VARST*23890./S(12)

```

```
000610      RETURN
000610      C      ERROR IN HOM ITERATION      SET IND TO -1
000611      900  IND=-1
000611      RETURN
000612      END
```

```

      SURROUTINE LFB (XP,FP,TX)
C      TX(1)      INITIAL GUESS
C      TX(2)      RATIO TO GET SECOND POINT
C      TX(3)      ZERO DEFINITION
C      TX(10)     COUNT OF NUMBER OF ITERATIONS
C                  SET TO ZERO ON SOLUTION
C                  SET TO NEGATIVE OF COUNT ON ERROR
C      FP        =FUNCTION(XP)
C      WHEN A SOLUTION IS FOUND, XP IS THE RDDT
C
C      ERROR EXITS OCCUR FOR
C      1. TOO MANY ITERATIONS, .GT.CNTMAX
C      2. TWO SUCESSIVE XP S OR FP S ARE EQUAL
000005      DIMENSION TX(10)
000005      DATA CNTMAX /1000./
000005      IF (TX(10).LE.0.) GO TO 1
000006      TX(10)=TX(10)+1.
000011      IF (TX(10)-3.) 2,3,4
C      ENTRY FIRST TIME THROUGH
000014      1 TX(10)=1.
000016      IF (TX(1).EQ.0) TX(1)=1.
000020      XP=TX(1)
C      GO GET F(XP)
000021      RETURN
C      ENTRY SECOND TIME THROUGH
000022      2 TX(9)=FP
000024      TX(8)=XP
000025      TX(5)=FP
000026      IF (ABS(FP).LT.TX(3)) GO TO 18
000030      XP=TX(1)*TX(2)
C      GO GET F(XP)
000031      RETURN
C      ENTRY THIRD TIME THROUGH
000032      3 TX(5)=FP
000034      TX(6)=XP
000035      TX(7)=FP
000036      IF (ABS(FP).LT.TX(3)) GO TO 18
000040      XP=TX(6)-TX(7)*(TX(6)-TX(8))/(TX(7)-TX(9))
C      GO GET F(XP)
000047      RETURN
C      ENTRY FOR FOURTH AND SUCCEEDING TIMES THROUGH
000050      4 IF (TX(10).GT.CNTMAX) GO TO 99
000054      TX(4)=XP
000055      TX(5)=FP
000056      T=TX(4)-TX(6)
000060      IF (T.EQ.0.) GO TO 99
000061      IF (ABS(FP).LT.TX(3)) GO TO 18
000063      R=TX(5)-TX(7)
000065      IF (R.EQ.0.) GO TO 99
000066      XP=TX(4)-TX(5)*(T/R)
000072      IF (TX(5)*TX(7).LT.0.) GO TO 11
000074      IF (TX(5)*TX(9).GE.0.) GO TO 11
000076      IF (XP.GT.TX(4)) GO TO 6
000102      IF (XP.GT.TX(8)) GO TO 10
000105      8 XP=TX(4)-TX(5)*(TX(4)-TX(8))/(TX(5)-TX(9))
000114      10 TX(7)=TX(5)
000116      TX(6)=TX(4)
C      GO GET F(XP)
000117      RETURN
000120      6 IF (XP.GT.TX(8)) GO TO 8

```

```
000124      GO TO 10
000124      11 TX(9)=TX(7)
000126      TX(8)=TX(6)
000127      GO TO 10
          C  HAVE FOUND A SOLUTION
000130      18 TX(10)=0.
000131      TX(1)=XP
000132      TX(4)=XP
000133      RETURN
          C  AN ERROR HAS OCCURED
          C  SET COUNT NEGATIVE AND EXIT
000134      99 TX(10)=-TX(10)
000136      RETURN
000136      END
```


APPENDIX C

THE GAMMA-LAW EQUATION OF STATE AND HUGONIOT TEMPERATURE PROGRAMS

The HOM equation of state used in FORTRAN SIN requires fits to the equation of state parameters for detonation products and condensed components. The FORTRAN BKW code⁸ produces the coefficients to the required fits using the Becker-Kistiakowsky-Wilson equation of state to describe the isentrope of the detonation products.

In this appendix we describe a code which will produce the coefficients to the required fits using a gamma-law equation of state to describe the isentrope of the detonation products. We also describe a code which calculates the single-shock Hugoniot temperatures using the technique of Walsh and Christian⁹ and produces the required fits for use in the FORTRAN SIN code.

The Gamma-Law Equation of State Program

A. The Nomenclature

- CJ Chapman-Jouguet
- D detonation velocity
- P pressure
- U_p particle velocity
- V volume
- V_o initial specific volume
- Z constant added to energies to make them positive
- γ gamma
- ρ_o initial density

B. The Equations

$$P_{CJ} V_{CJ}^{\gamma} = C$$

$$V_{CJ} = \frac{\gamma V_o}{(\gamma + 1)}$$

$$\ln P = \ln C + \gamma \ln V$$

$$I = \frac{P V}{\gamma - 1} - \frac{P_{CJ} V_{CJ}}{\gamma - 1} + \frac{P_{CJ}}{2}(V_o - V_{CJ}) + Z$$

$$U_p = \frac{P_{CJ}}{\alpha' \rho D} \left[1 + \alpha' - \left(\frac{P}{P_{CJ}} \right)^{\alpha'} \right] \text{ where } \alpha' = \frac{\gamma - 1}{2\gamma}$$

$$\alpha = \left[\frac{(\alpha + 1)}{\left(1 + \frac{d \ln D}{d \ln \rho} \right)} - 2 \right]^{-1}$$

$$3 = \frac{1 + \alpha}{\gamma}$$

C. The Input to the Code

		<u>1st card</u>
<u>Col.</u>	<u>Format</u>	
1-72	12A6	Alphabetic label for problem
		<u>2nd card</u>
<u>Col.</u>	<u>Format</u>	
1-18	E18.11	γ
19-36	E18.11	P _{CJ} (Mbar)
37-54	E18.11	ρ _o (g/cc)
		<u>3rd card</u>
<u>Col.</u>	<u>Format</u>	
1-18	E18.11	D
19-36	E18.11	d(ln D)/d(ln ρ)
37-54	E18.11	Z

D. The FORTRAN Code

```

PROGRAM GLAW (INPUT,OUTPUT,FILM,TAPE12=FILM)
C
GAMMA,PCJ,RHO,DETV,DDDR,Z ARE INPUT CONSTANTS
000003 DIMENSION P(100),E(100),V(100),ALGP(100),ALGV(100),ALNE(100)
1,ALNP(100),UP(100),FTE(100),ECOEF(5),ALNV(100),W(100)
2,DELY(100),SB(5),T(5),A(5,5),LABEL(12),ST(5)
000003 DATA PLFACT/+0.8/
000003 DATA PMAX /+1.0/
000003 DATA PMIN /+1.0E-4/
000003 DATA PUFAC T/+1.15/
000003 DATA AM /+0.4342944819/
000003 DATA ICHAR /044/
000003 DATA ICHAR2/063/
500 READ 900,LABEL
000011 READ 901,GAMMA,PCJ,RHO
000023 READ 901,DETV,DDDR,Z
000035 900 FORMAT (12A6)
000035 901 FORMAT (4E18.11)
000035 FITR=-GAMMA
000037 ALPHA= 1.0/((GAMMA+1.)/(1.+DDDR) - 2.)
000045 BETA= (1+ALPHA)/GAMMA
000050 FITR= -1./BETA
000051 VCJ= GAMMA/((GAMMA+1.)*RHO)
000054 C = PCJ*(VCJ**GAMMA)
000060 ALNC= ALOG(C)
000062 ALPHP= (GAMMA -1.)/(2.*GAMMA)
000065 ENCT = -PCJ*VCJ/(GAMMA-1.) + 0.5*PCJ*(1./RHO -VCJ) + Z
000077 P(1)=PCJ
000100 I=1
000101 K=1
000102 DO 10 J = 1,100
000103 20 ALNP(I)=ALOG(P(I))
000107 ALGP(I)=ALNP(I)*AM
000111 ALNV(I)=- (ALNP(I)-ALNC)/ GAMMA
000115 ALGV(I)= ALNV(I)*AM
000117 V(I)=EXP(ALNV(I))
000122 E(I)= (P(I)*V(I))/(GAMMA-1.)+ENCT
000130 ALNE(I) = ALOG(E(I))
000134 IF(P(I).GT. PCJ) GO TO 12
000140 UP(I)= PCJ/(RHO* ALPHP*DETV)*(1.+ALPHP-(P(I)/PCJ)**ALPHP)
000152 P(I+1) =P(I)*PLFACT
000154 IF(P(I+1).LT. PMIN) GO TO 11
000156 I=I+1
000160 K=K+1
000161 GO TO 20
000161 11 P(I+1) = PCJ*PUFACT
000164 GO TO 13
000164 12 P(I+1)=P(I)*PUFACT
000167 13 IF(P(I+1).GT.PMAX) GO TO 30
000173 I=I+1
000174 GO TO 20
000174 10 CONTINUE
000176 30 CALL PFTS(I,4,0,SIGMA,ALNP,ALNE,W,FTE,DELY,ECOEF,SB,T,ST,A)
000214 PRINT 911,LABEL
000222 PRINT 902,GAMMA,PCJ,RHO
000234 PRINT 903,DETV,DDDR,Z
000246 PRINT 904,ALPHA,BETA,ENCT
000260 PRINT 905,FITR,C
000270 PRINT 906,ALNC,FITB
000300 PRINT 907,(ECOEF(I),I=1,5)
000312 PRINT 908
000316 NOH=I-K

```

```

000320      M= I
000321      DO 40 L = 1,NOH
000322      FTE(M)=EXP(FTE(M))
000326      PRINT 909,P(M),V(M),E(M),FTE(M)
000342      M=M-1
000344      40 CONTINUE
000346      DO 50 L = 1,K
000350      FTE(L)=EXP(FTE(L))
000354      PRINT 910,P(L),V(L),E(L),FTE(L),UP(L)
000372      50 CONTINUE
000375      WRITE (12,911) LABEL
000402      WRITE (12,902) GAMMA,PCJ,RHO
000414      WRITE (12,903) DETV,DDDR,Z
000426      WRITE (12,904) ALPHA,BETA,ENCT
000440      WRITE (12,905) FITR,C
000450      WRITE (12,906) ALNC,FITB
000460      WRITE (12,907) (ECOE(I),I=1,5)
000472      WRITE (12,908)
000476      M=I
000500      DO 140 L = 1,NOH
000501      WRITE (12,909) P(M),V(M),E(M),FTE(M)
000514      M=M-1
000516      140 CONTINUE
000520      DO 150 L=1,K
000522      WRITE (12,910) P(L),V(L),E(L),FTE(L),UP(L)
000537      150 CONTINUE
C          GRAPH LOG PRESSURE VS VOLUME
000542      CALL ADV(1)
000543      CALL DGA (123,1023,0,900,-1.,+2.,+0.,-4.)
000553      CALL DLGLG
000554      CALL SLLOG
000555      CALL SRLG
000556      CALL PLOT (I,ALGV,1,ALGP,1,ICHAR,0)
000565      CALL PLOT (1,ALGV,1,ALGP,1,ICHAR2,0)
000574      CALL LINCNT (60)
000576      WRITE (12,900) LABEL
000604      WRITE (12,912)
C          GRAPH LOG PRESSURE VS PARTICLE VELOCITY
000610      CALL ADV(1)
000612      CALL DGA (123,1023,0,900,0.,+1.,+0.,-4.)
000622      CALL DLNLG(10)
000624      CALL SLLOG
000625      CALL SBLIN(10,2)
000627      CALL PLOT (K,UP,1,ALGP,1,ICHAR,1)
000636      CALL LINCNT (60)
000640      WRITE (12,900) LABEL
000646      WRITE (12,913)
000652      GO TO 500
000653      911 FORMAT (27H1 GAMMA LAW CALCULATION FOR,12A6)
000653      902 FORMAT (/,10H GAMMA IS ,1PE18.11,17H C-J PRESSURE IS ,1PE18.11,
112H DENSITY IS ,1PE18.11)
000653      903 FORMAT (/,24H DETONATION VELOCITY IS ,1PE18.11,18H DLND/DLN(RHO) I
IS ,1PE18.11,31H Z-CONSTANT ADDED TO ENERGY IS ,1PE18.11)
000653      904 FORMAT (/,19H COMPUTED ALPHA IS ,1PE18.11, 9H BETA IS ,1PE18.11,
123H ENERGY CONSTANT IS ,1PE18.11)
000653      905 FORMAT (/,11H LNT = Q + ,1PE18.11,30H LNV AND PV TO GAMMA IS
1 ,1PE18.11)
000653      906 FORMAT (/, 8H LN(P)= ,1PE18.11,3X,1PE18.11,5HLNV )
000653      907 FORMAT (/,8H LN(E)= ,1PE18.11,3X,1PE18.11,5HLNP ,1PE18.11,6HLNP*2
1 ,1PE18.11,6HLNP*3 ,1PE18.11,5HLNP*4)
000653      908 FORMAT (//,112H PRESSURE(MBAHS) VOLUME (CC/GM) ENERGY
1(MB-CC/GM) FIT ENERGY PARTICLE VELOCITY(CM/USEC))

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```
000653 909 FORMAT(4(3X.1PE18.11))
000653 910 FORMAT(5(3X.1PE18.11))
000653 912 FORMAT (59H THE PRESSURE-VOLUME ISENTROPE FOR GAMMA LAW ASSUMPT
1ION )
000653 913 FORMAT (46H THE PRESSURE -PARTICLE VELOCITY ISENTROPE )
000653 END
```

```

SURROUTINE PFTS(M,KM,IW,SIGMA,X,F2,W,Y,DELY,B,SB,T,ST,A)
C THIS IS A SPECIAL VERSION OF LA-PFTS FOR A MAX OF 100 POINTS
C AND A MAX OF FIFTH DEGREE FIT NO PRINTING OR LEGENDRE POLYNOMIALS
C ALSO WILL COMPUTE TILL GET THE FIT AND REQUIRES NO OUTPUT SUBROUTINE
C M = NO OF DATA POINTS
C KM = DEGREE OF FIT (MAX OF 5)
C IW = 0 FOR NO WEIGHTS AND EQUAL 1 FOR WEIGHTS
C SIGMA IS STANDARD DEVIATION COMPUTED
C X = X DATA INPUT ARRAY
C F2 = F2 DATA ARRAY INPUT
C W = WEIGHT DATA ARRAY INPUT SET EQUAL TO 1 IF IW IS 0
C Y = F COMPUTED FROM FIT USING X
C DELY = DIFFERENCE IN COMPUTED AND INPUT F
C H = COEFFICIENTS TO FIT
C SH = ESTIMATE OF ERRORS IN COEFFICIENTS TO FIT
C T = COEFFICIENTS TO ORTHOGONAL POLYNOMIALS
C ST = ERRORS IN T
C A = AREA USED BY CALCULATION
000017 DIMENSION S(5),X(100),F2(100),ST(5),SB(5),F(100),PM(100),P(100)
1 * B(5),DELY(100),W(100),A(5,5),T(5),Y(100)
000017 LL=0
000020 9 FM=0.0
000021 A(1,1)=1.0
000024 A(2,2)=1.0
000026 FBAR=0.0
000027 XBAR=0.0
000030 DO10I=1,M
000031 IF(IW)1009,1010,1009
000032 1010 W2=1.0
000034 W(I)=1.0
000037 GOTO1011
000040 1009 W2=SQRT(W(I))
000051 1011 FM=FM+W(I)
000055 F(I)=W2*F2(I)
000061 PM(I) = W2
000064 FBAR=FBAR+F(I)*PM(I)
000071 10 XBAR=XBAR+X(I)*PM(I)**2
000077 XBAR=XBAR/FM
000100 T(1)=FBAR/FM
000103 A(2,1)=-XBAR
000106 PXF=0.0
000107 PXP=0.0
000110 DO20I=1,M
000111 P(I)=(X(I)-XBAR)*PM(I)
000117 PXF=PXPF+P(I)*F(I)
000124 20 PXP=PXPF+P(I)*P(I)
000131 T(2)=PXPF/PXP
000134 PMXPM=FM
000136 S(1)=PMXPM
000137 KM=KM+1
000140 H(1)=T(1)*A(1,1)+T(2)*A(2,1)
000147 H(2)=T(2)*A(2,2)
000154 60 DO190K=2,KM
000156 IF(K-2)40,165,65
000160 40 STOP
000162 65 XXP=0.0
000163 XXP=XPXP+X(J)*P(J)
000164 H(K)=0.0
000167 DO70J=1,M
000171 XP=X(J)*P(J)
000175 XXP=XPXP+XP*P(J)

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000200      70  XPXPM=XPXPM+XP*PM(J)
000205          ALPHA=XPXP/PXP
000207          BETA=XPXPM/PMXPM
000211          PPXF=0.0
000212          PPXPP=0.0
000213          D090I=1,M
000214      80  PT=P(I)
000217      81  P(I)=X(I)*PT-ALPHA*PT-BETA*PM(I)
000230      82  PPXF=PPXF+P(I)*F(I)
000235      83  PPXPP=PPXPP+P(I)*P(I)
000241      90  PM(I)=PT
000246          T(K)=PPXF/PPXPP
000252          PMXPM=XPXP
000254          PXP=PPXPP
000254          A(K,1)=-ALPHA*A(K-1,1)-BETA*A(K-2,1)
000264          A(K,K-1)=A(K-1,K-2)-A(K-1,K-1)*ALPHA
000276          A(K,K)=1.0
000302          IF(K-3)150,150,110
000304      110 K1=K-2
000306          D0120I=2,K1
000310      120 A(K,I)=A(K-1,I-1)-ALPHA*A(K-1,I)-BETA*A(K-2,I)
000342      150 D0160I=1,K
000344      160 H(I)=B(I)+T(K)*A(K,I)
000364      165 SIG2=0.0
000365          D0180I=1,M
000367          Y(I)=POLY(X(I),K,R)
000404      175 DELY(I)=Y(I)-F2(I)
000413      180 SIG2=SIG2+(DELY(I)**2)*W(I)
000423          SIG2=SIG2/FLOAT(M-K)
000426          SIGMA=SQRT(SIG2)
000433          S(K)=PXP
000437          D0499I=1,K
000441      499 ST(I)=SIGMA/SQRT(S(I))
000457          D0501I=1,K
000461          SB(I)=0.0
000464          D0500J=1,K
000465      500 SB(I)=SB(I)+(A(J,I)*ST(J))**2
000507      501 SR(I)=SQRT(SB(I))
000522      190 CONTINUE
000524      220 KM=KM-1
000526          RETURN
000526          END

```

The Hugoniot Temperature Program

A. The Nomenclature

- C } coefficients to $U_s = C + S(U_p)$ used
- S } from P_o to switch pressure

- C1 } coefficients to $U_s = C1 + S1(U_p)$ used
- S1 } from switch pressure to maximum pressure

- C1 } coefficients for electronic term
- $C_{2\rho_o}$ } of Cowan equation of state
- $(C_{3\rho_o})^2$ }

- C_V heat capacity (cal/g-°C)
- K isothermal compressibility (Mbar⁻¹)
- P pressure
- T temperature (°K)
- T_o initial temperature (°K)
- T_V temperature (volts)
- U_p particle velocity
- U_s shock velocity
- V_o $1/\rho_o$
- α linear coefficient of expansion (°C⁻¹)
- ΔV volume increment ($\sim 1 \times 10^{-4}$) where $V_o - V_{smallest}/\Delta V < 5000$
- ρ density (g/cc)
- ρ_o initial density (g/cc)

B. The Equations

The following equation is solved

$$T = T_o e^{b(V_o - V)} + \frac{(V_o - V)P}{2C_V} + \frac{e^{-bV}}{2C_V} \times \int_{V_o}^V P e^{bV} [2 - b(V_o - V)] dV,$$

with

$$b = \frac{3\alpha}{K(C_V)}, \text{ using Simpson's rule.}$$

The input volume increment is used to compute a table of volumes which are used to compute pressures from

$$P = \frac{C^2(V_o - V)}{[V_o - S(V_o - V)]^2}.$$

The shock and particle velocities are found from

$$U_s = \frac{C + \sqrt{C^2 + 4SPV_o}}{2}$$

and

$$U_p = \frac{U_s - C}{S}.$$

The Cowan equation of state is

$$P = A + B\rho + C\rho^2 + D\rho^3 + E\rho^4 + (A1 + A2\rho)T_V + C_1 + \frac{(C_{2\rho_o})}{\rho} + \frac{(C_{3\rho_o})^2}{\rho^2} T_V^2,$$

where

$$\frac{3\alpha}{K} = A1 + A2\rho_o, \text{ and } A2 = \frac{1.447404}{(\text{Atomic weight})},$$

which is the form used in FORTRAN BKW.⁸

C. The Input to the Code

		<u>1st card</u>
<u>Col.</u>	<u>Format</u>	
1-72	12A6	Alphabetic label for problem
		<u>2nd card</u>
<u>Col.</u>	<u>Format</u>	
1-18	E18.11	C
19-36	E18.11	S
37-54	E18.11	ρ_o
55-72	E18.11	K
		<u>3rd card</u>
<u>Col.</u>	<u>Format</u>	
1-18	E18.11	α
19-36	E18.11	C_V
37-54	E18.11	T_o
55-72	E18.11	ΔV
		<u>4th card</u>
<u>Col.</u>	<u>Format</u>	
1-18	E18.11	Smallest P for temperature fit
19-36	E18.11	Largest P for temperature fit
		<u>5th card</u>
<u>Col.</u>	<u>Format</u>	
1-18	E18.11	Maximum pressure to be calculated
19-36	E18.11	C1
37-54	E18.11	S1
55-72	E18.11	Switch pressure

Note: If the two-phase feature is used, the temperature fit will include for the C1, S1 volumes only those temperatures which increase with decreasing volume.

6th card

Col.	Format	
1-18	E18.11	Atomic weight. If zero, no Cowan fit.
19-36	E18.11	C_1
37-54	E18.11	$C_2^{\rho_0}$
55-72	E18.11	$(C_3^{\rho_0})^2$

D. The FORTRAN Code

```

PROGRAM SEQS (INPUT,OUTPUT,FILM,TAPE12=FILM)
000007  DIMENSION V(5011),P(5010),EP(5010),T(500) ,US(500) ,UP(500)
1,TC(5),FITT(500),LAREL(12),TV(500),ALNT(500),ALNVT(500)
2,TCALC(500),W(500),DELY(500),SB(5),TZ(5),A(5.5),PV(500),PP(500)
3,CRHO(500),ZP(500),CCOEF(5),ST(5) ,TX(10)

000003  DATA DKV /+11605.6/
000003  DATA CALMB /23890./
000003  DATA ICHAR /055/
000003  DATA TX(1) /0./
000003  DATA TX(2) /1.01/
000003  DATA TX(3) /+1.0E-5/
000003  DATA TX(10) /0./
000003  366 READ 900.LABFL
000011  READ 901.C.S,RHO,AK
000025  READ 901.ALPHA,CP,TO,DELTV
000041  READ 901.TP1,TP2
000051  READ 901.AMAXP,C1,S1,SWP
000065  READ 901.ATWT,AC1,AC2,AC3
C      C,S,C1,S1 COEFFICIENTS TO US-UP FITS
C      SWITCH FROM C,S TO C1,S1 AT SWP -THE SWITCH PRESSURE
C      ALPHA = LINEAR COEFFICIENT OF EXPANSION
C      CP = HEAT CAPACITY IN CAL/GM/DEGC
C      AK = ISOTHERMAL COMPRESSIBILITY IN MBARS
C      RHO = DENSITY GM/CC
C      TO = INITIAL TEMPERATURE IN DEG KELVIN
C      DELTV = VOLUME INCREMENT APT 1 X 10-4
C      AMAXP = MAX PRESSURE
C      ATWT,AC1,AC2,AC3 COWAN EQ COEFFICIENTS SKIP IF ZERO
000101  V0=1./PH0
000103  Y=C
000105  Z=S
000106  I = 2
000107  BCV=CP/CALMB
000111  B= 3.*ALPHA/(AK*BCV)
000114  V(2) = V0-DELTV
000116  SWV = 0.
000117  DO 10 J = 2,5000
000121  11 P(I)=(Y+Y*(V0-V(I)))/((V0-Z*(V0-V(I)))*(V0-Z*(V0-V(I))))
000132  C      IF(P(I).GT. AMAXP) GO TO 13
FOR PHASE CHANGE USE CONSTANT PRESSURE
IF (P(I).LT.SWP.AND.Y.EQ.C1) P(I) = SWP
IF(P(I).LT. SWP) GO TO 12
000135  Y=C1
000147  Z=S1
000152  Y=C1
000153  Z=S1
000155  IF (SWV.FQ.0.) SWV = V(I)
000157  12 EP(I)=P(I)*EXP(B*V(I))*(2.-B*(V0-V(I)))
000173  V(I+1) = V(I)-DELTV
000175  I = I+1
000176  10 CONTINUE
000200  13 K=2
000201  M = I + 10
000203  DO 14 L = I,M
000204  P(L) = 0.
000205  14 CONTINUE
000207  I = I -1
000211  IT=1
000212  EP(1) = 0.
000213  PP(1) = 0.
000214  PV(1) = V0
000215  T(1) = TO
000217  US(1) = C

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000220      UP(1) = 0.
000221      Y=C
000222      Z=S
000223      SUM = 0.
000224      DO 20 J=1,I,10
000225  21 PP(K)=P(J+10)
000230      IF (PP(K).LT.+1.E-8) GO TO 30
000232      PV(K)=V(J+10)
000234      IF (PV(K).GT.SWV) GO TO 27
000237      Y=C1
000240      Z=S1
000241  27 SUM = SUM + (EP(J)+4.*EP(J+1)+7.*EP(J+2)+ 4.*EP(J+3
1)+2.*EP(J+4)+4.*EP(J+5)+2.*EP(J+6)+4.*EP(J+7)+2.* EP(J+8)+
24.*EP(J+9)+ FP(J+10))
      T(K)= T0*EXP(B*(V0-PV(K)))+(V0-PV(K))*PP(K)/(2.*BCV) -
1*((EXP(-H*PV(K)))/(2.*BCV))*SUM*DELT V/3.
000271  22 US(K)=(Y+ SQRT(Y*Y + 4.*PP(K)*V0*Z))*0.5
000321      UP(K)=(US(K) -Y)/Z
000334      IF (T(K).LT.T(K-1)) GO TO 23
000337      IF (PP(K).GT.TP1.AND.PP(K).LT.TP2) GO TO 24
000341      GO TO 23
000351  24 ALNVT(IT)=ALOG(PV(K))
000355      ALNT(IT)=ALOG(T(K))
000361      IT=IT+1
000363  23 K=K+1
000365  20 CONTINUE
000367  30 IT =IT-1
000371      K=K-1
000372      HIP = PP(K)
000374      HIV = V0
000375      HIT = T(K)
000377      HIUS = US(K)
000400      HIUP = UP(K)
000402      CALL PFTS (IT,4.0,SIGMA,ALNVT,ALNT,W,TCALC,DELY,TC,SB,TZ,ST,A)
000417      PRINT 902,LABEL
000425      PRINT 903,C,S,SWP
000437      IF (SWP.GT.AMAXP) GO TO 41
000443      PRINT 904,SWP,AMAXP,C1,S1
000456  41 PRINT 905,RHO
000464      PRINT 906,AK
000472      PRINT 907,ALPHA
000500      PRINT 908,T0
000506      PRINT 909,CP
000514      PRINT 910,DELTV
000522      PRINT 911,TP1,TP2
000532      PRINT 912, (TC(I),I=1,5)
000544      PRINT 913
000550      DO 42 I=1,K
000552      PRINT 914, PV(I),PP(I),T(I),US(I),UP(I)
000567  42 CONTINUE
000572      PRINT 915
000575      DO 43 I=1,IT
000577      ALNVT(I)= EXP(ALNVT(I))
000603      ALNT(I)= EXP(ALNT(I))
000610      TCALC(I)=EXP(TCALC(I))
000615      PRINT 916, ALNVT(I), ALNT(I),TCALC(I)
000627  43 CONTINUE
000632      WRITE(12,902) LABEL
000637      WRITE(12,903) C,S,SWP
000651      IF (SWP.GT.AMAXP) GO TO 51
000655      WRITE(12,904) SWP,AMAXP,C1,S1
000670  51 WRITE(12,905) RHO

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000674      WRITE(12,906) AK
000704      WRITE(12,907) ALPHA
000712      WRITE(12,908) TO
000720      WRITE(12,909) CP
000726      WRITE(12,910) DELTV
000734      WRITE(12,911) TP1,TP2
000744      WRITE(12,912) (TC(I),I=1,5)
000756      WRITE(12,913)
000762      DO 52 I=1,K
000764      WRITE(12,914) PV(I),PP(I),T(I),US(I),UP(I)
001001      52 CONTINUE
001004      WRITE(12,915)
001007      DO 53 I=1,IT
001011      WRITE(12,914) ALNVT(I), ALNT(I),TCALC(I)
001022      53 CONTINUE
C          PRESSURE -VOLUME PLOT
001025      CALL ADV(1)
001026      CALL DGA(123,1023,0.900,0.,HIV,HIP,0.)
001036      CALL DLNLN (20,20)
001040      CALL SLLIN (20,2)
001042      CALL SRLIN (20,2)
001044      CALL PLOT (K,PV,1,PP,1,ICHA,1)
001053      CALL LINCNT (60)
001055      WRITE (12,923)
001061      WRITE (12,900) LAREL
C          TEMPERATURE-VOLUME PLOT
001067      CALL ADV(1)
001071      CALL DLNLN (20,20)
001073      CALL DGA(123,1023,0.900,0.,HIV,HIT,0.)
001103      CALL SLLIN (20,1)
001105      CALL SRLIN (20,2)
001107      CALL PLOT (K,PV,1,T,1,ICHA,1)
001114      CALL LINCNT (60)
001120      WRITE (12,924)
001124      WRITE (12,900) LAREL
C          PRESSURE-PARTICLE VELOCITY PLOT
001132      CALL ADV(1)
001134      CALL DGA(123,1023,0.900,0.,1.0,1.5,0.)
001144      CALL DLNLN (20,20)
001146      CALL SLLIN (20,2)
001150      CALL SRLIN (20,2)
001152      CALL PLOT (K,UP,1,PP,1,ICHA,1)
001161      CALL LINCNT (60)
001163      WRITE (12,925)
001167      WRITE (12,900) LAHFL
C          SHOCK-PARTICLE VELOCITY PLOT
001175      CALL ADV(1)
001177      CALL DGA(123,1023,0.900,0.,HIUP,HIUS,0.)
001207      CALL DLNLN (20,20)
001211      CALL SLLIN (20,2)
001213      CALL SRLIN (20,2)
001215      CALL PLOT (K,UP,1,US,1,ICHA,1)
001224      CALL LINCNT (60)
001226      WRITE (12,926)
001232      WRITE (12,900) LAREL
C          CALCULATE RESIDUAL TEMPERATURES
001240      600 TX(1) = V0
001242      DO 601 I = 1,K
001243      605 CALL LFR(VC,F,TX)
001246      IF (TX(1)) 602,603,604
001250      604 F = -VC + V0 + V0*3.*ALPHA*(T(I))*EXP(B*(PV(I)-VC11-TO))
001265      GO TO 605

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C      USE US FOR RESIDUAL VOLUME AND UP FOR RESIDUAL TEMPERATURE
001265 603 US(I) = VC
001267      UP(I)=T(I) *EXP(B*(PV(I)-VC))
001276 601 CONTINUE
001300      GO TO 700
001301 602 PRINT 930
001305      GO TO 701
001306 700 PRINT 931
001312      PRINT 900,LABEL
001320 701 PRINT 932
001324      DO 702 I = 1,K
001326      PRINT 914,PV(I),PP(I),T(I),US(I),UP(I)
001343 702 CONTINUE
001346      WRITE (12,931)
001351      WRITE (12,900) LABEL
001357      WRITE (12,932)
001363      DO 703 I=1,K
001365      WRITE (12,914) PV(I),PP(I),T(I),US(I),UP(I)
001402 703 CONTINUE
001405      IF(ATWT.LT.0.1) GO TO 366
C      COWAN FIT
001407 100 DO 101 I=1,K
001411      TV(I)= T(I)/DKV
001413      CRHO(I)= 1./PV(I)
001415 101 CONTINUE
001417      A2=1.447404/ATWT
001421      A1=-A2*CRHO +3.0*ALPHA*DKV/AK
001426      DO 102 I=1,K
001430      ZP(I)= PP(I)-(A1+A2*CRHO(I))*TV(I) -(AC1+AC2/CRHO(I)+AC3*AC3/
1(CRHO(I)*CRHO(I))*TV(I)*TV(I)
001446 102 CONTINUE
001450      CALL PFTS (K,4.0,SIGMA,CRHO,ZP,W,FITT,DELY,CCOEF,SB,TZ,ST,A)
001466      PRINT 917
001472      PRINT 918,ATWT
001500      PRINT 919, (CCOEF(I),I=1,5)
001512      PRINT 920, A1,A2
001522      PRINT 921, AC1,AC2,AC3
001534      PRINT 922
001540      DO 103 I=1,K
001542      PRINT 914 ,ZP(I),FITT(I)
001551 103 CONTINUE
001554      WRITE (12,917)
001557      WRITE (12,918) ATWT
001565      WRITE (12,919) (CCOEF(I),I=1,5)
001577      WRITE (12,920) A1,A2
001607      WRITE (12,921) AC1,AC2,AC3
001621      WRITE (12,922)
001625      DO 104 J=1,K
001627      WRITE (12,914) ZP(I),FITT(I)
001636 104 CONTINUE
001641      GO TO 366
001641 900 FORMAT (12A6)
001641 901 FORMAT (4E18.11)
001641 902 FORMAT (41H1 SOLID EQUATION OF STATE CALCULATION FOR.12A6)
001641 903 FORMAT (/, 6H US = .1PE18.11,3H + .1PE18.11,17H S FROM PO TO .
11PF18.11,10H MEGABARS )
001641 904 FORMAT (/, 6H FROM .1PE18.11,13H MEGABARS TO .1PF18.11,29H MEGABAR
1S USED THE FIT US = .1PE18.11,3H + .1PE18.11, 3H S )
001641 905 FORMAT (/,24H THE INITIAL DENSITY IS .1PE18.11, 7H GM/CC )
001641 906 FORMAT (/,24H THE COMPRESSIBILITY IS .1PE18.11)
001641 907 FORMAT (/,40H THE LINEAR COEFFICIENT OF EXPANSION IS .1PE18.11)
001641 908 FORMAT (/,27H THE INITIAL TEMPERATURE IS.1PE18.11)

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001641 909 FORMAT (//.22H THE HEAT CAPACITY IS ,1PE18.11)
001641 910 FORMAT (//.25H THE VOLUME INCREMENT IS ,1PE18.11)
001641 911 FORMAT (//.32H THE TEMPERATURE FIT IS BETWEEN ,1PF18.11, 5H AND .
11PF18.11.10H MEGABARS )
001641 912 FORMAT (//.8H LN(T)= ,1PE18.11.3X.1PE18.11.5H LNV ,1PE18.11.6H LNV*2
1 ,1PE18.11.6H LNV*3 ,1PE18.11.5H LNV*4)
001641 913 FORMAT (//.110H VOLUME IN CC/GM PRESSURE IN MEGABARS TE
TEMPERATURE DEG K SHOCK VELOCITY PARTICLE VELOCITY)
001641 914 FORMAT (5(4X.1PE18.11))
001641 915 FORMAT ( 66H1 VOLUME IN CC/GM INPUT TEMPERATURE FI
1T TEMPERATURE )
001641 917 FORMAT (41H1 THE COWAN EQUATION OF STATE COEFFICIENTS )
001641 918 FORMAT (//.21H THE ATOMIC WEIGHT IS ,1PE18.11)
001641 919 FORMAT (//.6H P = ,1PE18.11.3X.1PE18.11.5HRHO ,1PE18.11.6HRHO*2 ,
11PE18.11.6HRHO*3 ,1PE18.11.6HRHO*4 )
001641 920 FORMAT (//.5H ( ,1PE18.11.3X.1PE18.11.23H RHO ) TEMPERATURE )
001641 921 FORMAT (//.5H ( ,1PE18.11.3X.1PE18.11, 6H /RHO ,1PE18.11.24H/RHO*
12 )*TEMPERATURE*2 )
001641 922 FORMAT (//.40H PRESSURE CALC FIT PRESSURE )
001641 923 FORMAT ( 40H PRESSURE-VOLUME HUGONIOT FOR )
001641 924 FORMAT ( 40H TEMPERATURE- VOLUME HUGONIOT FOR )
001641 925 FORMAT ( 50H PRESSURE -PARTICLE VELOCITY HUGONIOT FOR )
001641 926 FORMAT ( 50H SHOCK VELOCITY-PARTICLE VELOCITY HUGONIOT FOR )
001641 930 FORMAT(44H1 AN ERROR OCCURRED IN RESIDUAL TEMP CALC )
001641 931 FORMAT(44H1 RESIDUAL TEMPERATURE AND DENSITIES FOR )
001641 932 FORMAT (//.114H HUGONIOT VOLUME HUGONIOT PRESSURE HU
IGONIOT TEMPERATURE RESIDUAL VOLUME RESIDUAL TEMPERATURE )
001641 END

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SUBROUTINE LFR (XP,FP,TX)
C   TX(1)   INITIAL GUESS
C   TX(2)   RATIO TO GET SECOND POINT
C   TX(3)   ZFRO DEFINITION
C   TX(10)  COUNT OF NUMBER OF ITRATIONS
C           SET TO ZFRO ON SOLUTION
C           SET TO NEGATIVE OF COUNT ON ERROR
C   FP      =FUNCTION(XP)
C           WHEN A SOLUTION IS FOUND, XP IS THE ROOT
C
C           ERROR EXITS OCCUR FOR
C           1. TOO MANY ITERATIONS. .GT.CNTMAX
C           2. TWO SUCESSIVE XP S OR FP S ARE EQUAL
000005   DIMENSTON TX(10)
000005   DATA CNTMAX /1000./
000005   IF (TX(10).LE.0.) GO TO 1
000006   TX(10)=TX(10)+1.
000011   IF (TX(10)-3.) 2,3,4
C   ENTRY FIRST TIME THROUGH
000014   1 IF (TX(1).EQ.0.) TX(1)=1.
000016   TX(10)=1.
000020   XP=TX(1)
C   GO GET F(XP)
000021   RETURN
C   ENTRY SECOND TIME THROUGH
000021   2 TX(9)=FP
000023   TX(8)=XP
000024   TX(5)=FP
000025   IF (ABS(FP).LT.TX(3)) GO TO 18
000027   XP=TX(1)*TX(2)
C   GO GET F(XP)
000030   RETURN
C   ENTRY THIRD TIME THROUGH
000031   3 TX(5)=FP
000033   TX(6)=XP
000034   TX(7)=FP
000035   IF (ABS(FP).LT.TX(3)) GO TO 18
000037   XP=TX(6)-TX(7)*(TX(6)-TX(8))/(TX(7)-TX(9))
C   GO GET F(XP)
000046   RETURN
C   ENTRY FOR FOURTH AND SUCCEEDING TIMES THROUGH
000047   4 IF (TX(10).GT.CNTMAX) GO TO 99
000053   TX(4)=XP
000054   TX(5)=FP
000055   T=TX(4)-TX(6)
000057   IF (T.EQ.0.) GO TO 99
000060   IF (ABS(FP).LT.TX(3)) GO TO 18
000062   R=TX(5)-TX(7)
000064   IF (R.EQ.0.) GO TO 99
000065   XP=TX(4)-TX(5)*(T/R)
000071   IF (TX(5)*TX(7).LT.0.) GO TO 11
000073   IF (TX(5)*TX(9).GE.0.) GO TO 11
000075   IF (XP.GT.TX(4)) GO TO 6
000101   IF (XP.GT.TX(8)) GO TO 10
000104   R=XP-TX(4)-TX(5)*(TX(4)-TX(8))/(TX(5)-TX(4))
000113   10 TX(7)=TX(5)
000115   TX(8)=TX(4)
C   GO GET F(XP)
000116   RETURN
000117   5 IF (XP.GT.TX(8)) GO TO 8
000123   GO TO 10

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```
000123      11 TX(9)=TX(7)
000125      TX(8)=TX(6)
000126      GO TO 10
          C  HAVE FOUND A SOLUTION
000127      18 TX(10)=0.
000130      TX(1)=XP
000131      TX(4)=XP
000132      RETURN
          C  AN ERROR HAS OCCURED
          C  SET COUNT NEGATIVE AND EXIT
000133      99 TX(10)=-TX(10)
000134      RETURN
000135      END
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SUBROUTINE PFTS(M,KM,IW,SIGMA,X,F2,W,Y,DELY,B,SB,T,ST,A)
C THIS IS A SPFCIAL VERSION OF LA-PFTS FOR A MAX OF 500 POINTS
C AND A MAX OF FIFTH DEGREE FIT NO PRINTING OR LEGENDRE POLYNOMIALS
C ALSO WILL COMPUTE TILL GET THE FIT AND REQUIRES NO OUTPUT SUBROUTINE
C M = NO OF DATA POINTS
C KM = DEGREE OF FIT (MAX OF 5)
C IW = 0 FOR NO WEIGHTS AND EQUAL 1 FOR WEIGHTS
C SIGMA IS STANDARD DEVIATION COMPUTED
C X = X DATA INPUT ARRAY
C F2 = F2 DATA ARRAY INPUT
C W = WEIGHT DATA ARRAY INPUT SET EQUAL TO 1 IF IW IS 0
C Y = F COMPUTED FROM FIT USING X
C DELY = DIFFERENCE IN COMPUTED AND INPUT F
C B = COEFFICIENTS TO FIT
C SB = ESTIMATE OF ERRORS IN COEFFICIENTS TO FIT
C T = COEFFICIENTS TO ORTHOGONAL POLYNOMIALS
C ST = ERRORS IN T
C A = AREA USED BY CALCULATION
000017 DIMENSION S(5),X(500),F2(500),ST(5),SB(5),F(500),PM(500),P(500)
1 * B(5),DELY(500),W(500),A(5,5),T(5),Y(500)
000017 LL=0
000020 9 FM=0.0
000021 A(1,1)=1.0
000024 A(2,2)=1.0
000026 FBAR=0.0
000027 XBAR=0.0
000030 DO10I=1,M
000031 IF(IW)1009,1010,1009
000032 1010 W2=1.0
000034 W(I)=1.0
000037 GOTO1011
000040 1009 W2=SQRT(W(I))
000051 1011 FM=FM+W(I)
000055 F(I)=W2*F2(I)
000061 PM(I) = W2
000064 FBAR=FBAR+F(I)*PM(I)
000071 10 XBAR=XBAR+X(I)*PM(I)**2
000077 XBAR=XBAR/FM
000100 T(1)=FBAR/FM
000103 A(2,1)=-XBAR
000106 PXF=0.0
000107 PXP=0.0
000110 DO20I=1,M
000111 P(I)=(X(I)-XBAR)*PM(I)
000117 PXF=PXF+P(I)*F(I)
000124 20 PXP=PXP+P(I)*P(I)
000131 T(2)=PXG/PXP
000134 PMXPM=FM
000136 S(1)=PMXPM
000137 KM=KM+1
000140 B(1)=T(1)*A(j,1)+T(2)*A(2,1)
000147 B(2)=T(2)*A(2,2)
000154 60 DO190K=2,KM
000156 IF(K-2)40,165,65
000160 40 STOP
000162 65 XPP=0.0
000163 XPPM=0.0
000164 B(K)=0.0
000167 DO70J=1,M
000171 XP=X(J)*P(J)
000175 XPP=XPP+XP*P(J)

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000200      70  XPPM=XPPM+XP*PM(J)
000205      ALPHA=XPP/PXP
000207      BETA=XPPM/PMXP
000211      PPXF=0.0
000212      PPXP=0.0
000213      D090I=1.M
000214      80  PT=P(I)
000217      81  P(I)=X(I)*PT-ALPHA*PT-BETA*PM(I)
000230      82  PPXF=PPXF+P(I)*F(I)
000235      83  PPXP=PPXP+P(I)*P(I)
000241      90  PM(I)=PT
000246      T(K)=PPXF/PPXP
000252      PMXP=XP
000254      PXP=PPXP
000254      A(K,1)=-ALPHA*A(K-1,1)-BETA*A(K-2,1)
000264      A(K,K-1)=A(K-1,K-2)-A(K-1,K-1)*ALPHA
000277      A(K,K)=1.0
000303      IF(K-3)150,150,110
000305      110 K1=K-2
000307      D0120I=2.K1
000311      120 A(K,I)=A(K-1,I-1)-ALPHA*A(K-1,I)-BETA*A(K-2,I)
000344      150 D0160I=1.K
000346      160 B(I)=R(I)+T(K)*A(K,I)
000366      165 SIG2=0.0
000367      D0180I=1.M
000371      Y(I)=POLY (X(I),K,R)
000406      175 DELY(I)=Y(I)-F2(I)
000415      180 SIG2=SIG2+(DELY(I)**2)*W(I)
000426      SIG2=SIG2/FL0AT(M-K)
000430      SIGMA=SQRT(SIG2)
000436      S(K) = PXP
000442      D0499I=1.K
000444      499 ST(I)=SIGMA/SQRT(S(I))
000462      D0501I=1.K
000464      SB(I)=0.0
000467      D0500J=1.K
000470      500 SB(I)=SB(I)+(A(J,I)*ST(J))**2
000512      501 SB(I)=SQRT(SB(I))
000525      190 CONTINUE
000527      220 KM=KM-1
000531      RETURN
000531      ENO

```



```
000005      FUNCTION POLY (X,N,A)
000005      DIMENSION A(2)
000005      Y=A(N)
000007      DO 1 I=2,N
000010      J=N-I+1
000012      1 Y=A(J)+Y*X
000020      POLY =Y
000021      RETURN
000022      END
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